

Joint Research Centre

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Fast transient numerical simulations with EUROPLEXUS

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Directorate for Space Security and Migration

Safety and Security of Buildings

www.ec.europa.eu/jrc

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 - Explicit time integration
 - EUROPLEXUS introduction
- Fluid-Structure Interaction (FSI)
- Failure / Erosion / Debris
- Mesh Adaptivity
- Combustion

Introduction (JRC)

Terrorism



Paris 2015, 12 fatalities



Madrid 2004, 191 fatalities



[www.timesofisrael.com]

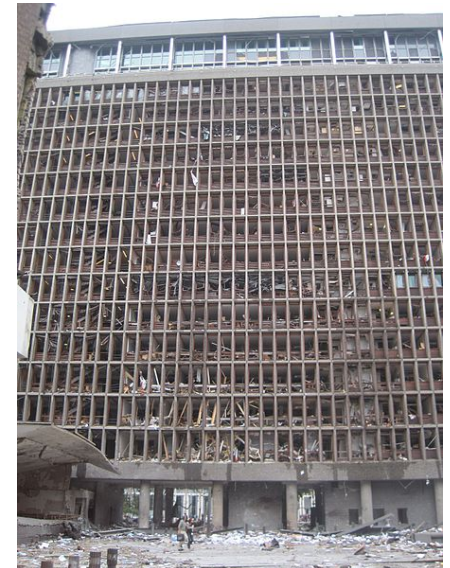
Ankara 2015, 102 fatalities



London 2005, 50 fatalities



New York 2001, 2993 fatalities



[wikipedia]

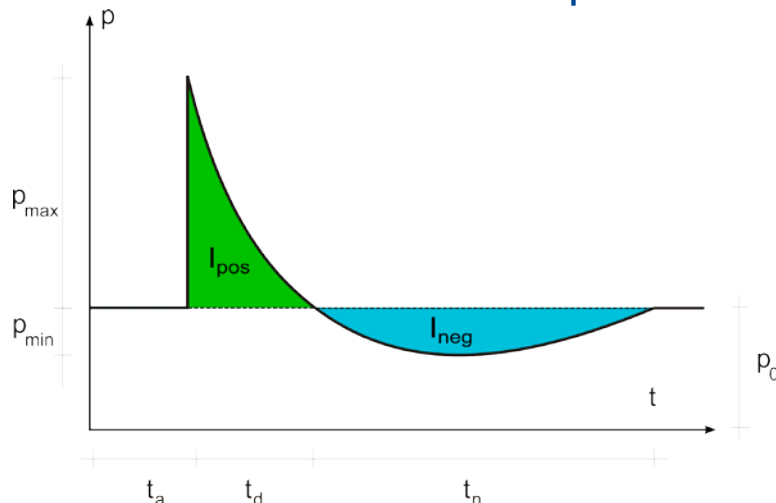
Norway 2011, 77 fatalities

Explosions, Air Blast Waves

- Contact detonations → local failure
- Air blast waves → in general structural failure

Air blast pressure depends on

- the distance
- the size of the explosive



[EMI]



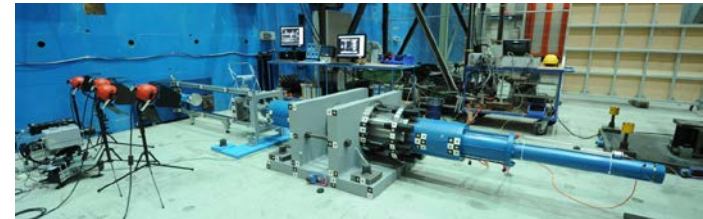
[wikipedia]

Norway 2011, 77 fatalities

Experiments → Model Calibration → Simulations

→ ELSA labs

- Experimental setup for fast dynamic testing of materials
- Experimental setup for testing small structures under blast loading

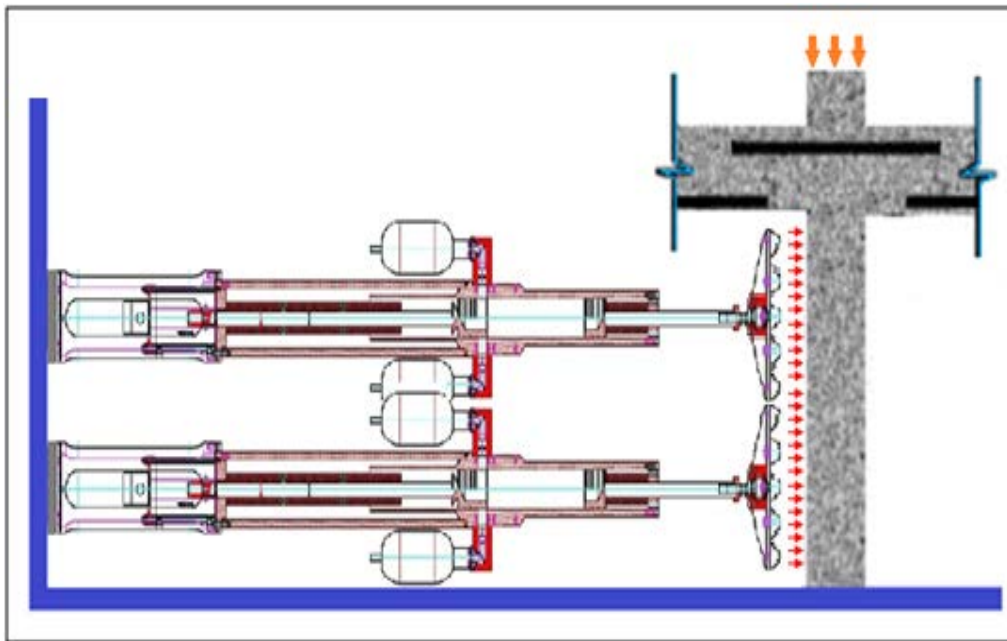


→ Simulation tool **EUROPLEXUS**, developed by JRC and CEA

- Explicit finite element code for fast dynamic response of structures (explosions, impacts, crashes, etc.)
- Specialized in modelling of Fluid-Structure Interaction phenomena
- Experience in simulation of safety problems

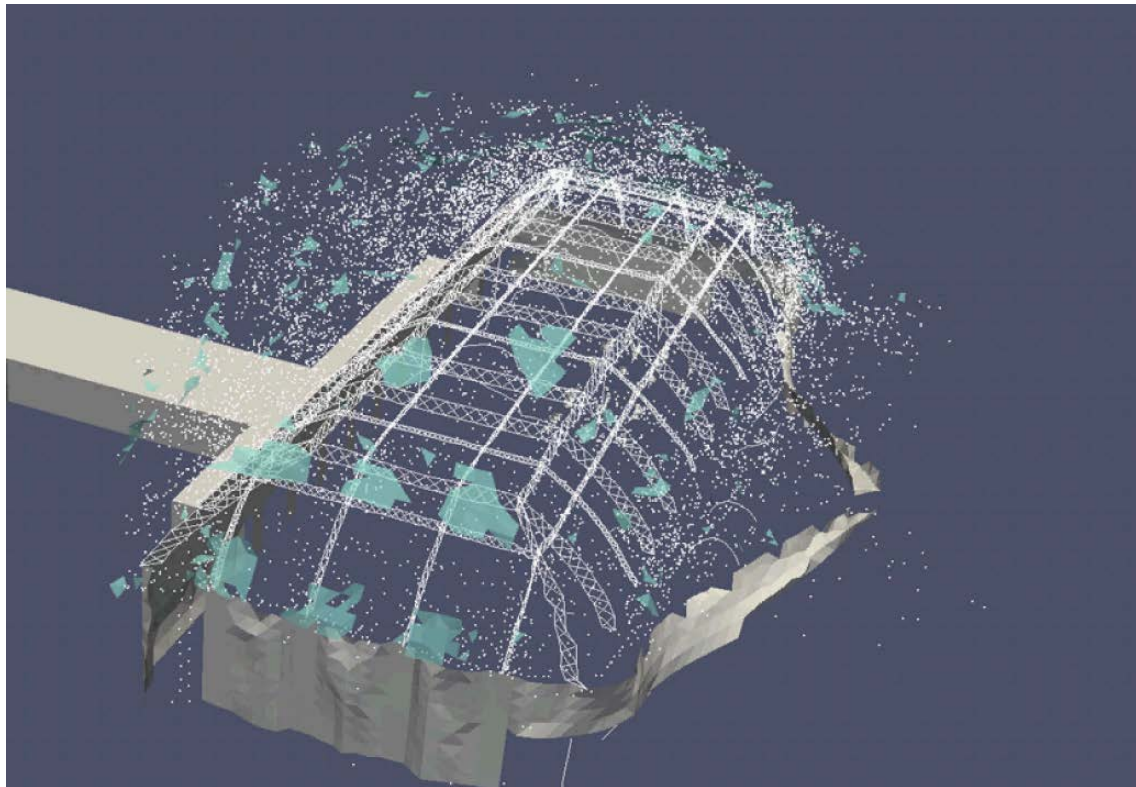
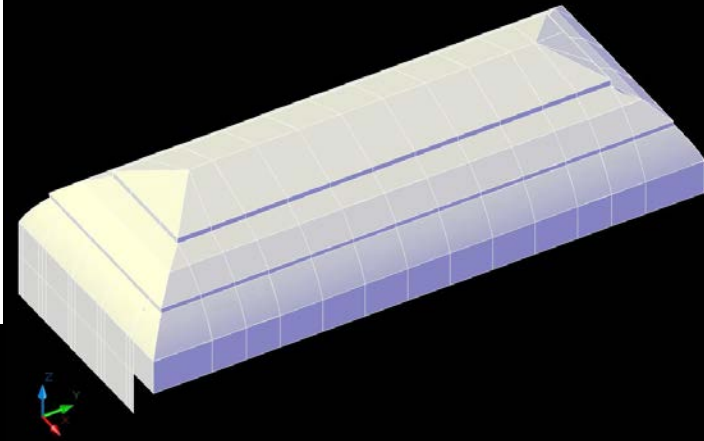


Blast Simulator



Time: 0.0000 sec

Explosion inside a station

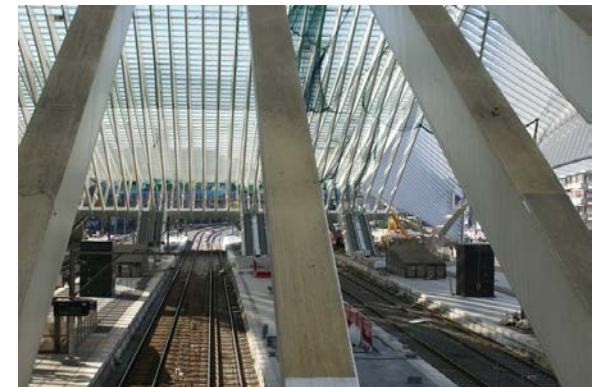
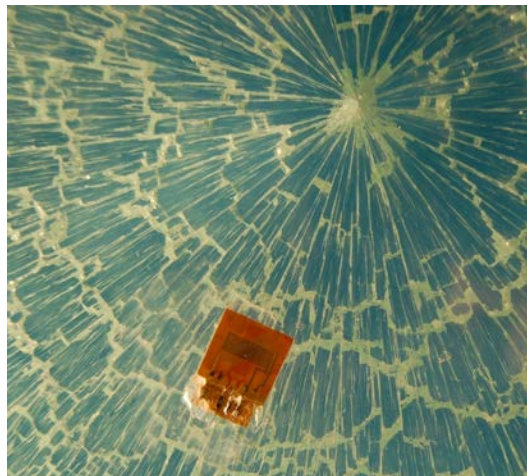
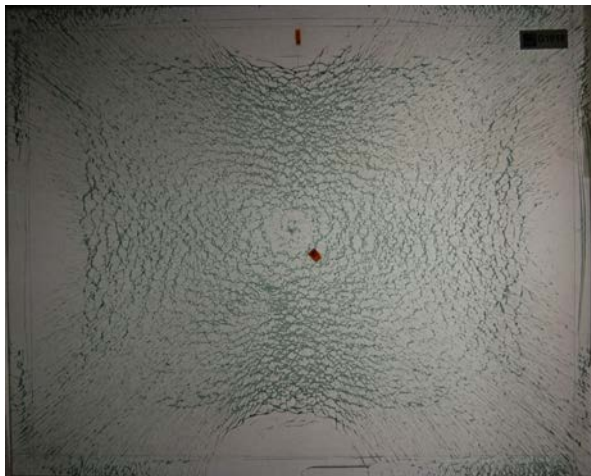


Air blast loading of glass

- Modern architecture uses very often big glass surfaces
- Glass most fragile part of a structure
- Failure results in splinters
- Air blast makes them fly into the building
- Laminated glass could help to hold them together



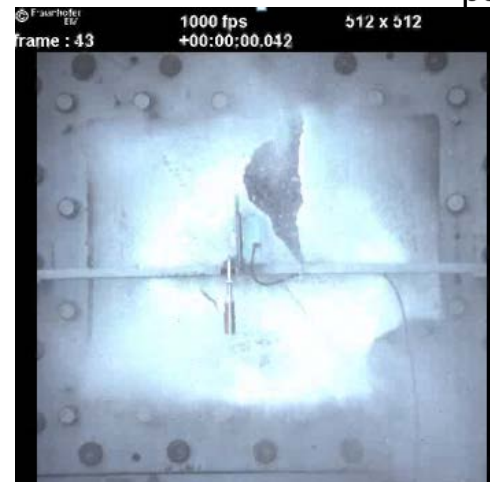
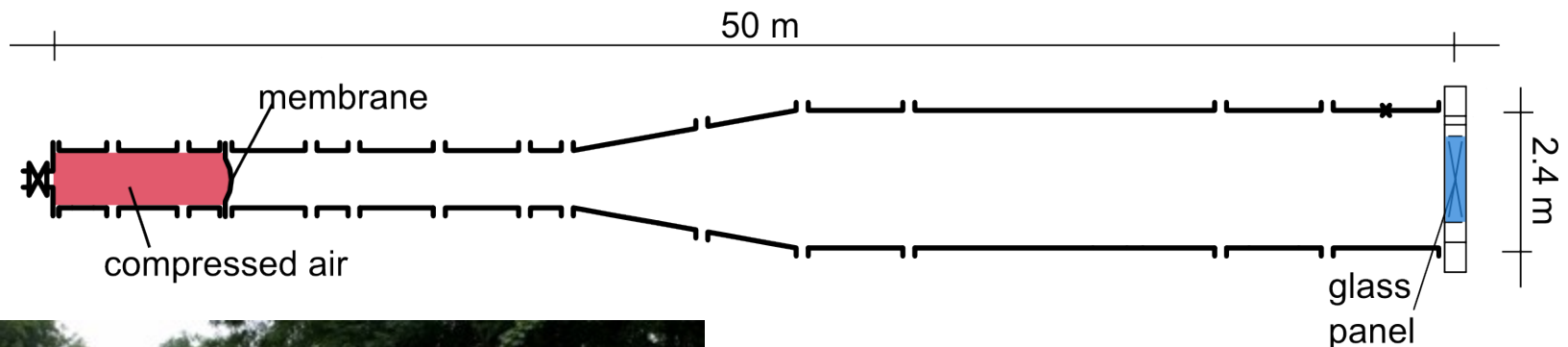
European Parliament,
Strasbourg



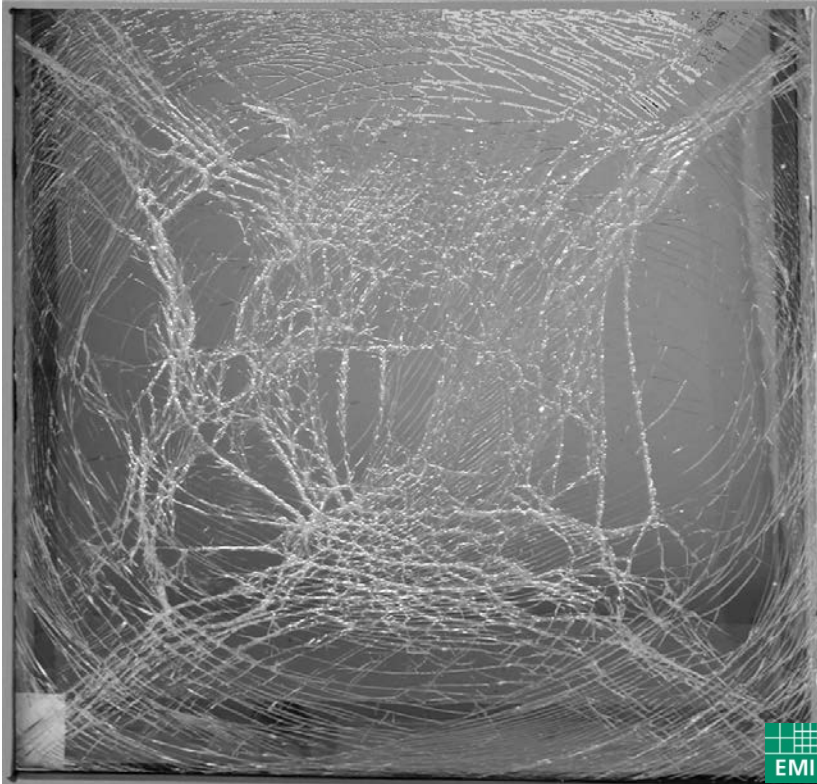
Railway station Liège

Experiments

- Full scale test at Ernst-Mach-Institut, Fraunhofergesellschaft, Germany
- Shock tube using compressed air, laminated glass 14 mm thickness
- Overpressures: 82 – 150 kPa (430 – 2300 kg TNT; 40-55 m distance)

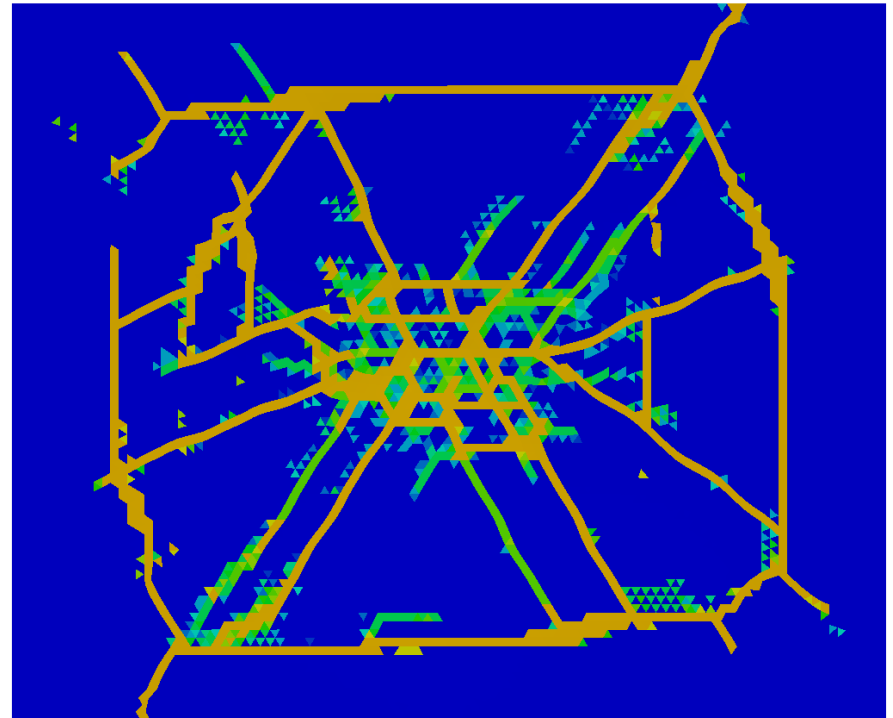


Numerical simulation: crack pattern



[Kranzer et al. 2005]

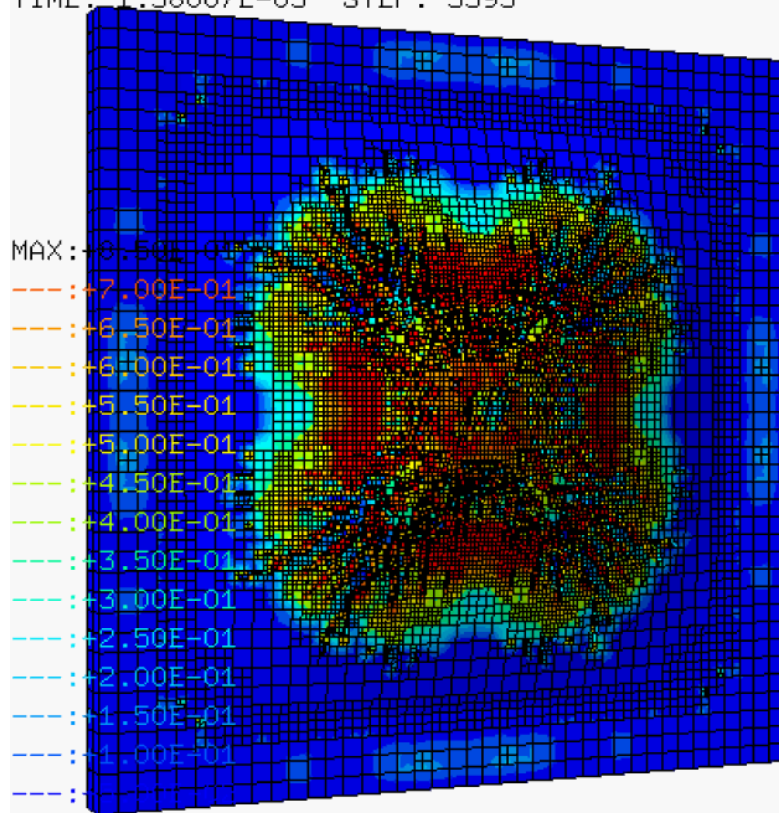
Experiment



Numerical failure

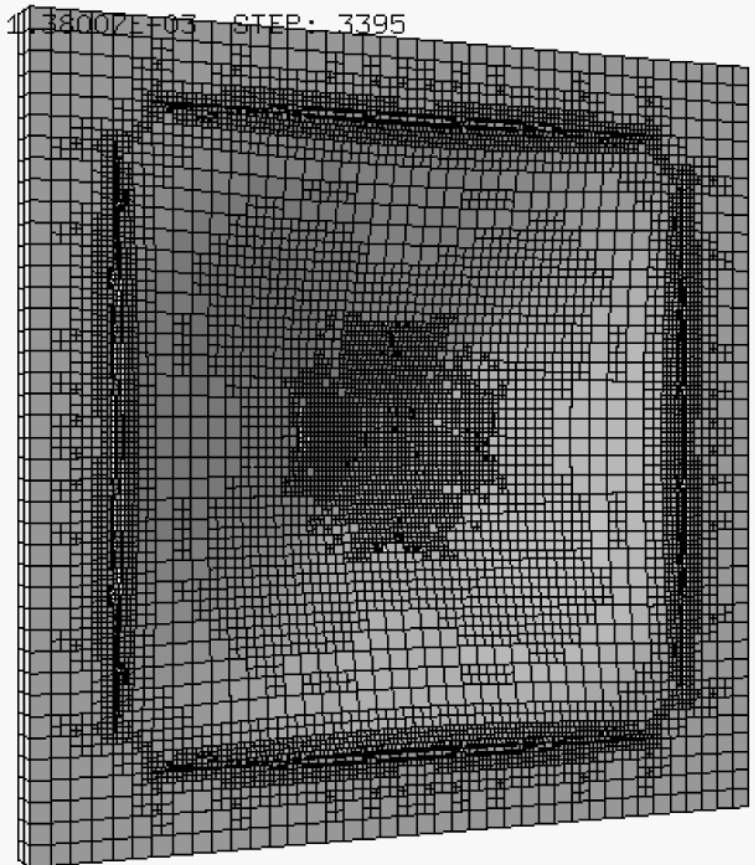
Adaptivity (concrete)

DPDC01
TIME: 1.38007E-03 STEP: 3395

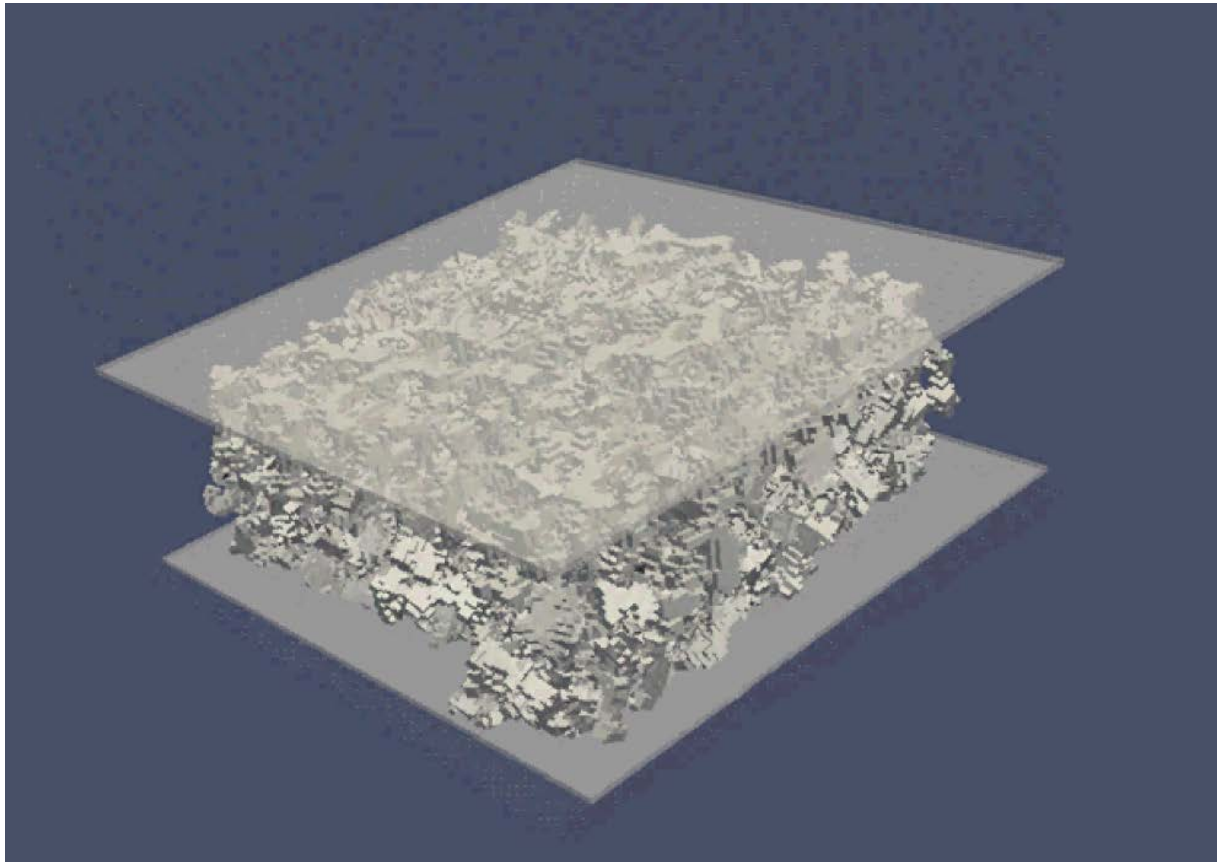


MIN: +0.00E+00
ECRO 13 [N/A]

DPDC01
TIME: 1.38007E-03 STEP: 3395



Aluminium foam



Metal tube



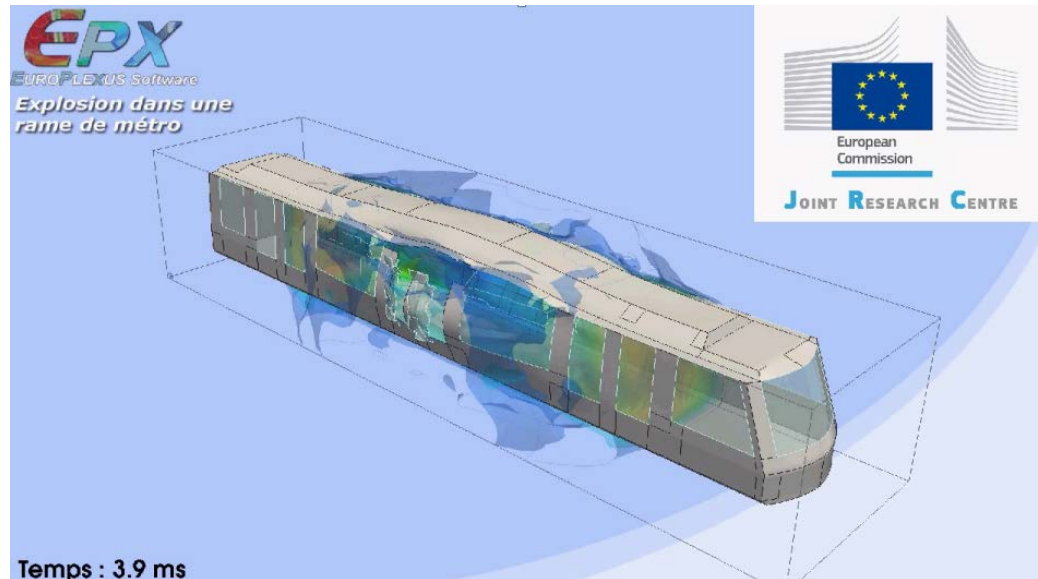
Soft targets

- Unarmored or undefended target (wikipedia)
- Explosions in urban terrain



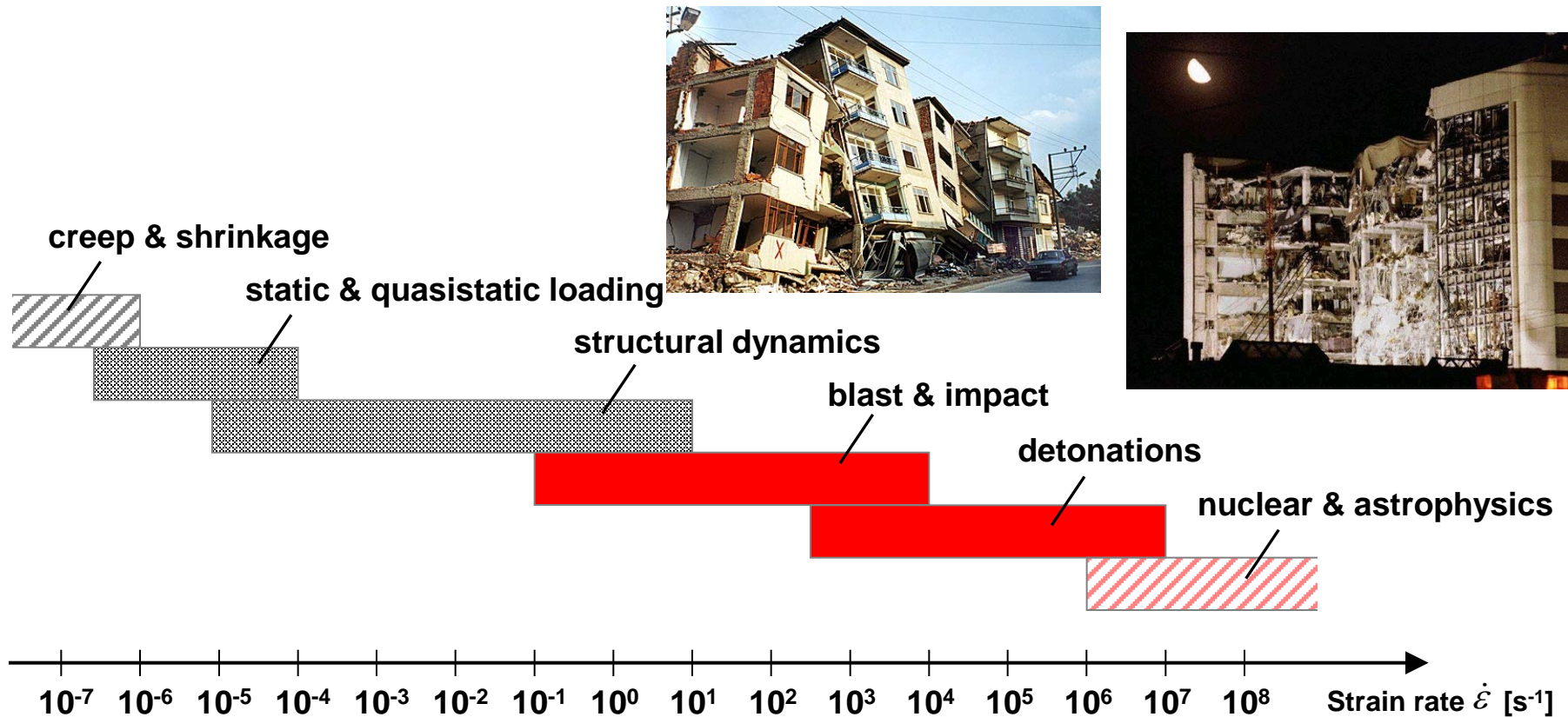
[www.timesofisrael.com]

Ankara 2015, 102 fatalities



Explicit time integration

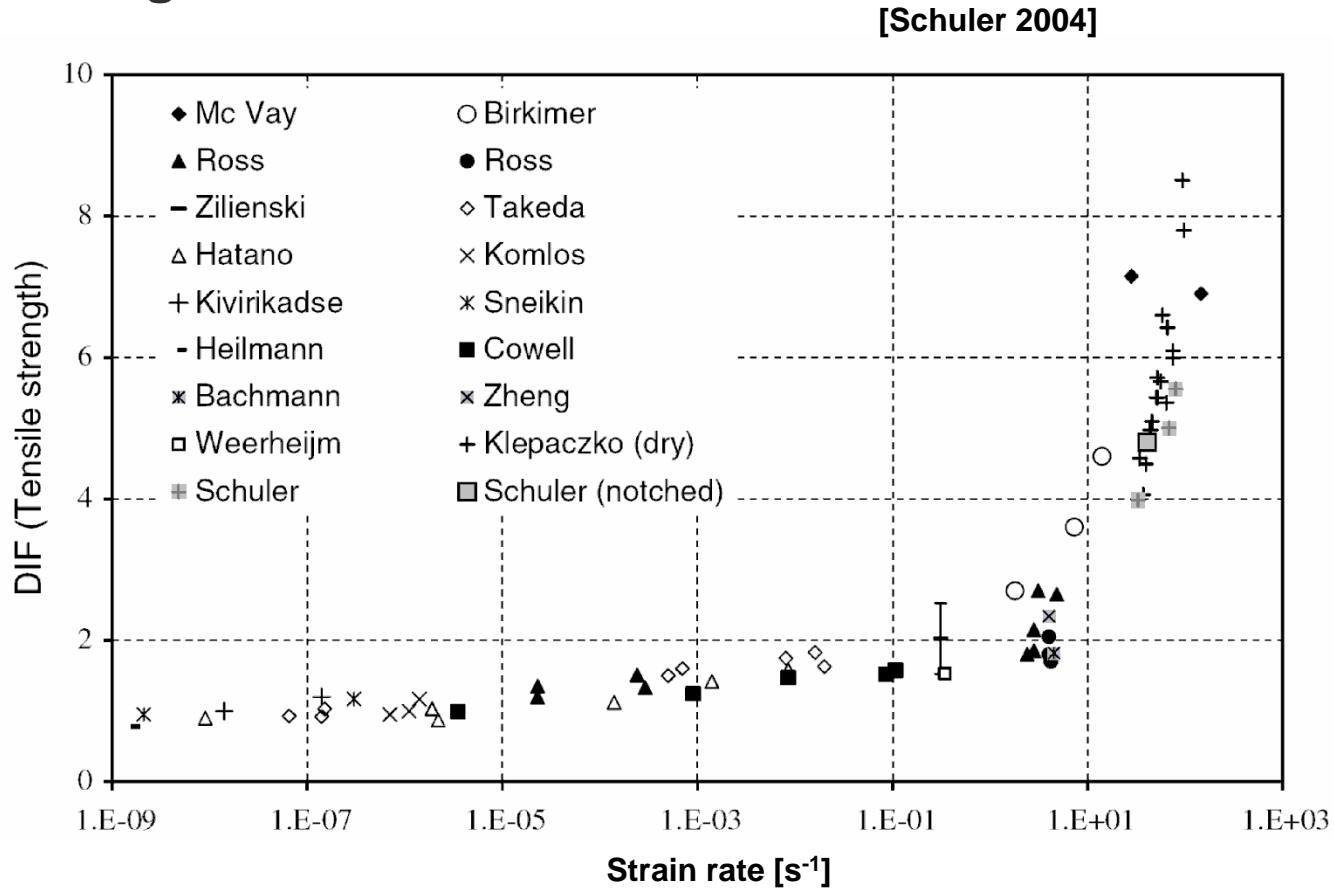
Dynamic loadings - Classification



time – strain rate – high pressures – wave propagation

Strain rate behaviour of concrete

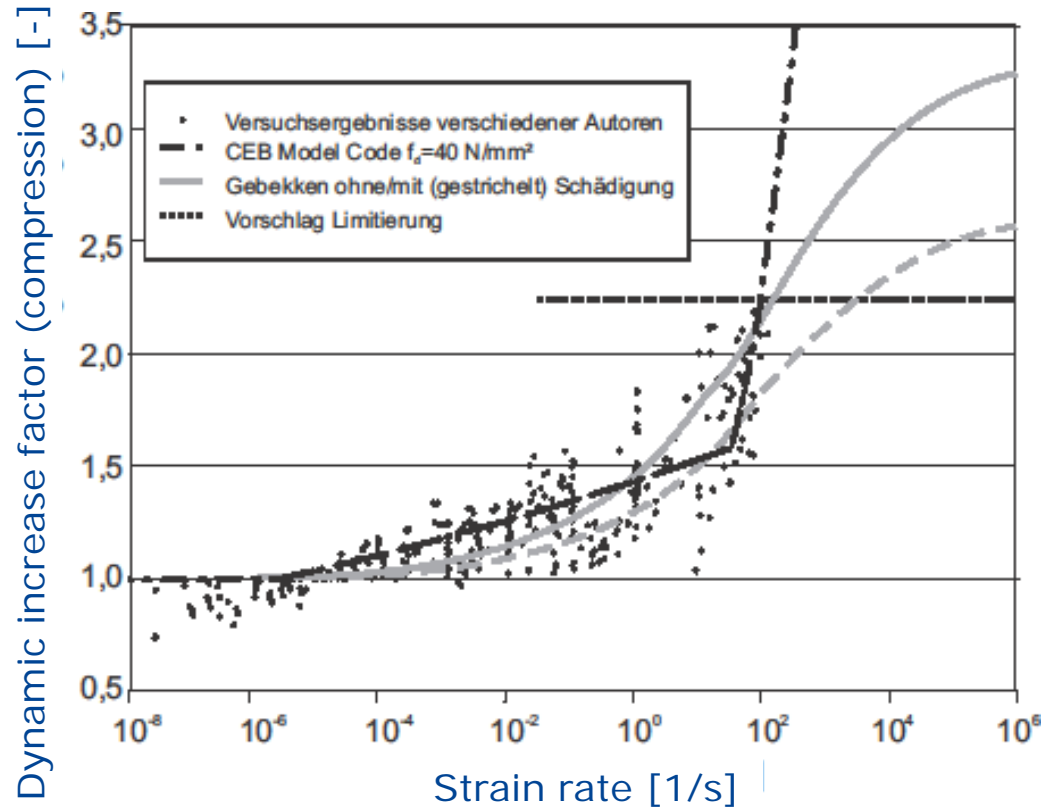
Tensile strength



Strain rate behaviour of concrete

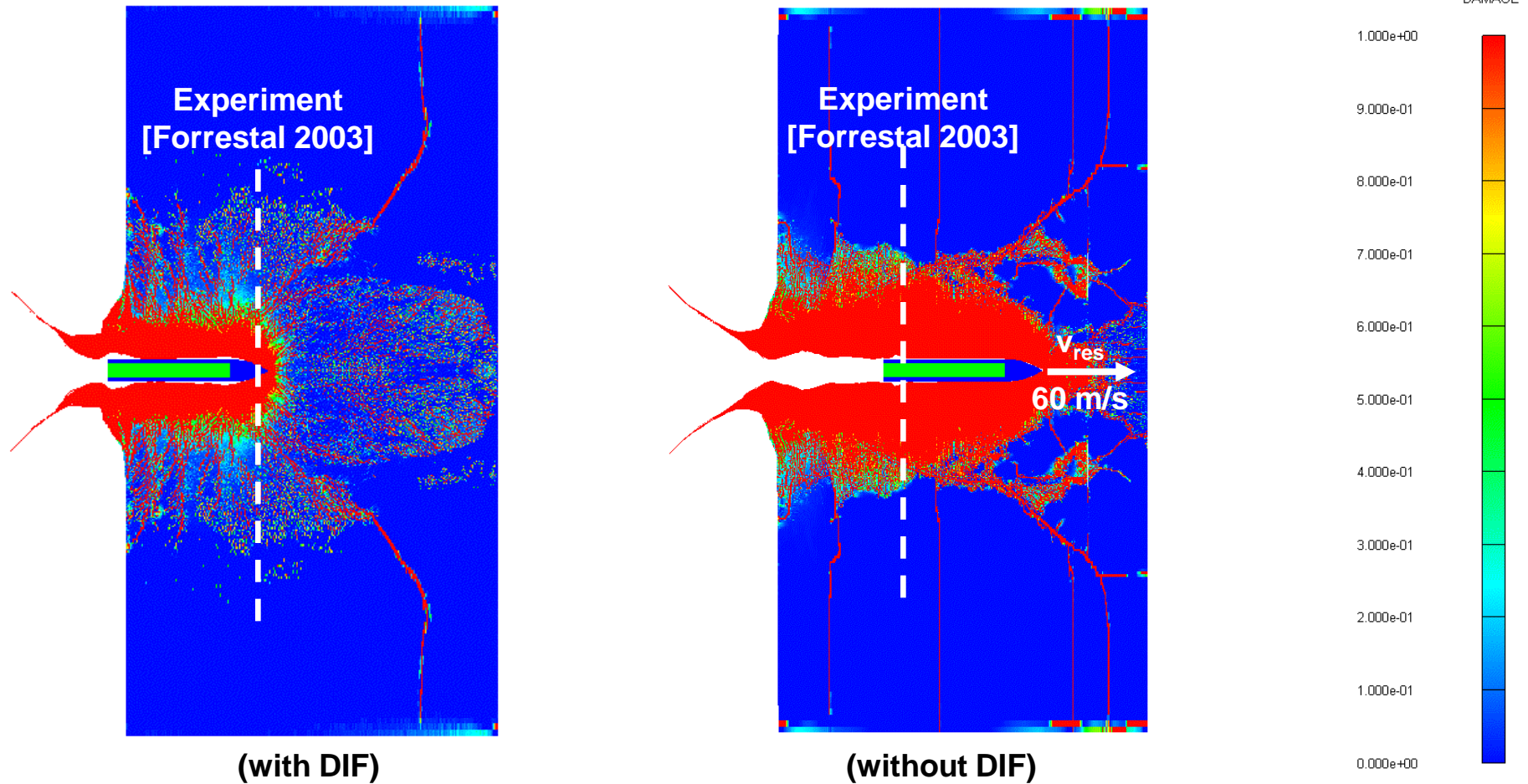
Compressive strength

[Larcher 2007]



Material modelling

Strain rate effects – example: penetration of a concrete target



Dynamic equations

Newtonian mechanics

$$F=ma$$

$$\mathbf{Ma} + \mathbf{Cv} + \mathbf{Ku} = \mathbf{f}$$

Newmark

$$\mathbf{M}\mathbf{a} + \mathbf{C}\mathbf{v} + \mathbf{K}\mathbf{u} = \mathbf{f}$$

Assumption for velocity and displacement:

$$\mathbf{v}_{n+1} = \mathbf{v}_n + \Delta t[(1 - \gamma)\mathbf{a}_n + \gamma\mathbf{a}_{n+1}]$$

$$\mathbf{u}_{n+1} = \mathbf{u}_n + \Delta t\mathbf{v}_n + \frac{\Delta t^2}{2}[(1 - 2\beta)\mathbf{a}_n + 2\beta\mathbf{a}_{n+1}]$$

$$\mathbf{M}\mathbf{a}_{n+1} + \mathbf{C}\mathbf{v}_{n+1} + \mathbf{K}\mathbf{u}_{n+1} = \mathbf{f}_{n+1}$$

$$(\mathbf{M} + \gamma\Delta t\mathbf{C} + \beta\Delta t^2\mathbf{K})\mathbf{a}_{n+1} = \mathbf{f}_{n+1} - \mathbf{C}\tilde{\mathbf{v}}_{n+1} - \mathbf{K}\tilde{\mathbf{u}}_{n+1}$$

$$\gamma = \frac{1}{2}, \beta = \frac{1}{4}$$

$$\tilde{\mathbf{v}}_{n+1} = \mathbf{v}_n + \Delta t(1 - \gamma)\mathbf{a}_n$$

$$\tilde{\mathbf{u}}_{n+1} = \mathbf{u}_n + \Delta t\mathbf{v}_n + \frac{\Delta t^2}{2}(1 - 2\beta)\mathbf{a}_n$$



Explicit Time Integration

$$\beta = 0 \quad \gamma = \frac{1}{2}$$

$$\mathbf{a}_{n+1} = \mathbf{a}_n + \gamma \Delta \mathbf{a}$$

$$\mathbf{v}_{n+1} = \mathbf{v}_n + \Delta t [(1 - \gamma) \mathbf{a}_n + \gamma \mathbf{a}_{n+1}]$$

$$\mathbf{u}_{n+1} = \mathbf{u}_n + \Delta t \mathbf{v}_n + \frac{\Delta t^2}{2} [(1 - 2\beta) \mathbf{a}_n + 2\beta \mathbf{a}_{n+1}]$$

$${}^t \mathbf{a} = \frac{1}{\Delta t^2} ({}^{t-\Delta t} \mathbf{u} - 2 {}^t \mathbf{u} + {}^{t+\Delta t} \mathbf{u})$$

$$\frac{1}{\Delta t^2} \mathbf{M} {}^{t+\Delta t} \mathbf{u} = {}^t \mathbf{f}_{int} - {}^t \mathbf{f}_{ext} - \left(\mathbf{K} - \frac{2}{\Delta t^2} \mathbf{M} \right) {}^t \mathbf{u} - \frac{1}{\Delta t^2} \mathbf{M} {}^{t-\Delta t} \mathbf{u}$$

Solution at $t+\Delta t$ is done with the values at $t \rightarrow$ balance at time t

No inversion of a matrix

Procedure can be done on element base

Explicit Time Integration

Forward directed method

Advantage

- No equilibrium iterations necessary
- No inversion of the stiffness matrix
- Procedure can be done on element base → MPI

→ Short calculation time

Literature: Belytschko

Disadvantage

- Critical time step length

$$\Delta t \leq \alpha_c \frac{d_{\min}}{c}$$

$$0,8 \leq \alpha_c \leq 0,98$$

Explicit Time Integration

Many steps but small steps

- Results from all steps cannot be stored
- Equilibrium is not given every time step
- Balance must not be given in a certain time step:
contact, erosion ... are easier or possible
- Boundary conditions needs special care (shown later on)
- Time step size is critical – check the results



fast transient dynamics • fluid-structure interaction • multiphysics

EUROPLEXUS introduction

EUROPLEXUS History

Development in cooperation between JRC, CEA

Major partner: EDF, ONERA

CASTEM/PLEXUS – CEA (Saclay, France): 1978

Collaboration EU: JRC (Ispra, Italy)/ CEA: 1982

- FLUID-STRUCTURE interaction
- ALE method
- PLEXIS-3C : 1986

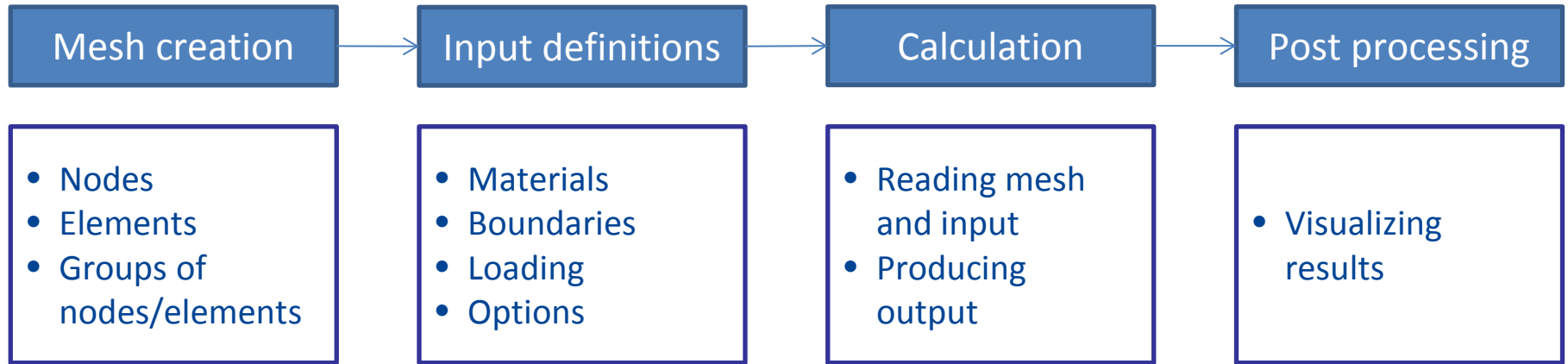
EUROPLEXUS project

- Merging of codes 1999
- CEA – JRC

EUROPLEXUS Licenses / principles

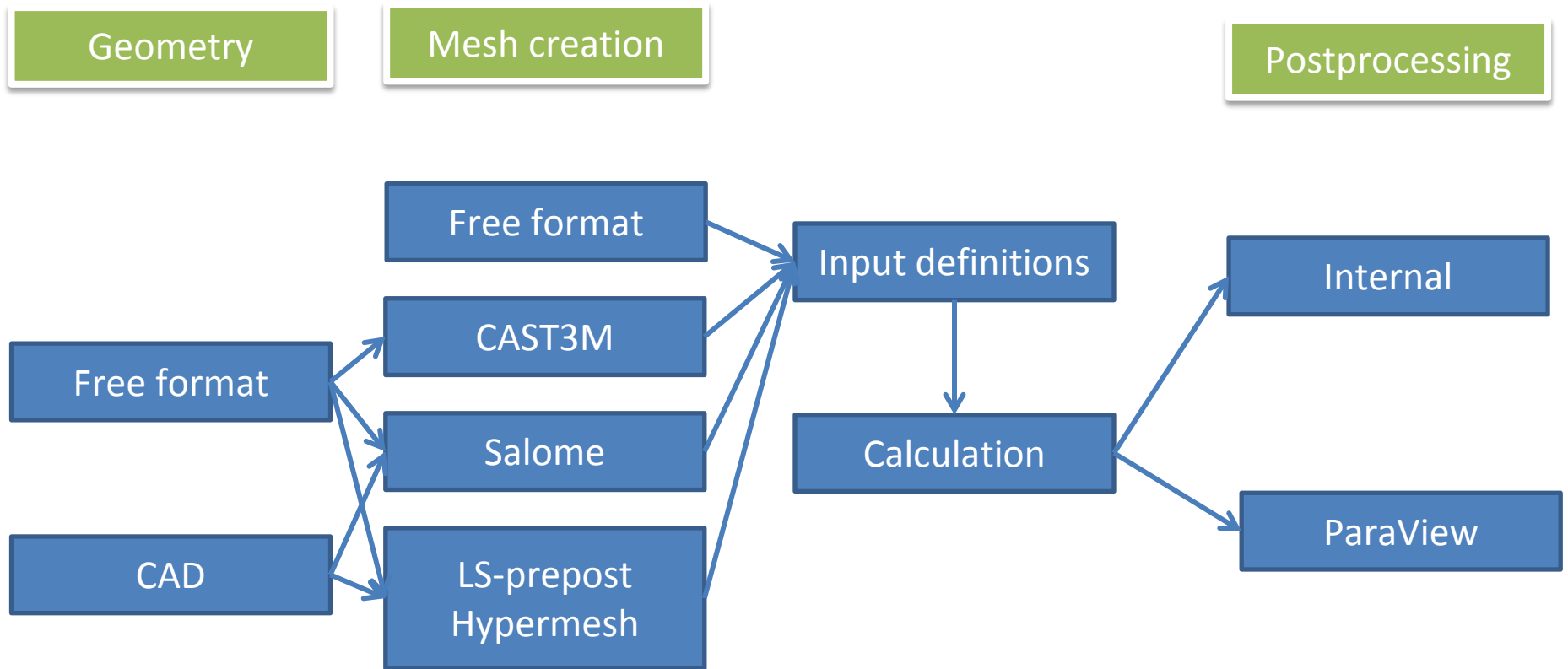
- Export control
- Distribution by CEA (<http://www-epx.cea.fr/>)
- Light version (20k stru and 200k fluid), downloadable
- Education and Research: free after signing a contract, also parts of the source for further developments
- Commercial license

General workflow



- Mesh and inputs separated
- Calculation: exe

Detailed workflow



Preprocessors

Objective: Create mesh (nodes, elements, groups)

.msh

CAST3M



Salome



- FE software (CEA), freely available
- Mesh created by defining points, lines, surfaces etc. (script language)
- Very powerful but difficult to learn

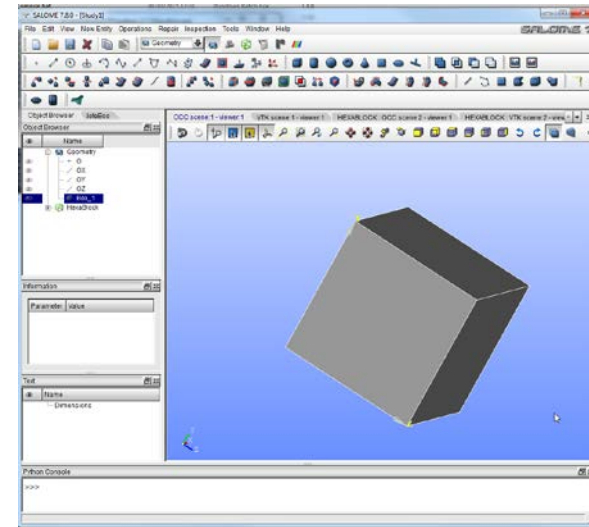
- generic platform for Pre- and Post-Processing (open-source)

.k

LS-prepost

Hypermesh

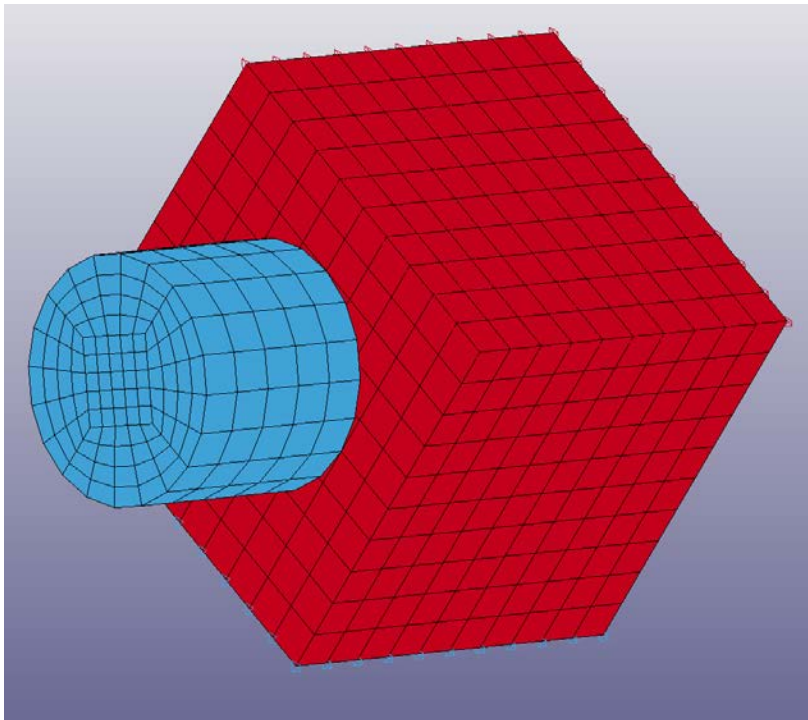
- Generic LS-DYNA k-file input



LS-prepost

- Free
- Easy to learn
- Also tool to check k-files

➔ Excuse: How to create mesh with Is-prepost



Organization of k-file

The following keywords are supported

- *NODE
- *ELEMENT_SHELL
- *ELEMENT_SOLID
- *SET_NODE_LIST
- *SET_NODE_ELEM

Postprocessors

Objective: visualize results (displacements, stress,...)



- Very powerful but more difficult to learn

- generic platform for Post-Processing (open-source)

Files

.epx EUROPLEXUS input file
.k LS-DYNA mesh file
.msh CASTEM mesh file
.listing General outputs
.std Error messages
.log Logging (one line per step)

EPX-Input file

- *.epx file
- Using groups of keywords (primary keywords)
- Creating epx-file: running through the groups

A	DIME	PROBLEM TYPE AND DIMENSIONING
B	GEOM	MESH AND GRID MOTION
C	COMP	GEOMETRIC COMPLEMENTS
C1	MATE	MATERIALS
D	LINK	LINKS
E	INIT	FUNCTIONS AND INITIAL CONDITIONS
F	CHAR	LOADS
G	ECRI	PRINTOUT AND STORAGE OF RESULTS
H	OPTI	OPTIONS
I	CALC	TRANSIENT CALCULATION DEFINITION
ED		POST-TREATMENT BY EUROPLEXUS
O		INTERACTIVE COMMANDS
V		The built-in OpenGL Graphical Visualizer

Manual

http://europlexus.jrc.ec.europa.eu/public/manual_pdf/manual.pdf

(development version!)

or including in the distribution

Contains

- Getting started
 - Keyword groups (A-...)
 - Bibliography
 - Index
- Hands on!



EUROPLEXUS

A Computer Program for the Finite Element Simulation of Fluid-Structure
Systems under Transient Dynamic Loading

USER'S MANUAL



JRC99891

Commissariat à l'énergie atomique
Direction de l'énergie nucléaire
Département de Modélisation des Systèmes et Structures
Service des Etudes Mécaniques et Thermiques
Laboratoire d'Etudes de Dynamique

European Commission
Joint Research Centre
Directorate for Space, Security and Migration
Safety and Security of Buildings

Units

- User defines the units in a coherent way
- Some exceptions for some materials / models
- SI units strongly recommended (m, kg, s, K)

First file!

```
1 impact0 !title of the problem
2 ECHO !Output on the console
3 KFIL !mesh file definition (k-file with name impact0.k)
4 TRID LAGR !3d structural calculation
5 GEOM Q4GS PART 1 PART 2 TERM !element definition
6 COMP EPAI 2 LECT PART 1 TERM !Thickness of part 1
7      5 LECT PART 2 TERM !Thickness of part 2
8 MATE LINE RO 7800 YOUNG 2.E11 NU 0.3 !Material definition: linear material
9      LECT PART 1 PART 2 TERM !for parts 1 and 2
10 LINK COUP !Links (coupled)
11      BLOQ 123 LECT NSET 4 NSET 5 TERM !Boundaries
12 INIT VITE 3 -110 LECT PART 2 TERM !Initial conditions
13 ECRI FICH PVTX TFRE 1.0E-3 !Output as paraview
14      VARI DEPL !Output variable displacement
15 OPTI NOTE LOG 1 !Log file written each step
16      CSTA 0.5 !Stability step
17 CALC TINI 0.0 TFIN 100.E-3 !Start and end time of calculation
18 FIN
19
```

Groups not covered here:

DIME, CHAR

ParaView



- open source
- multiple-platform
- interactive, scientific visualization
- client–server architecture
- built on top of the Visualization Toolkit (VTK) libraries
- www.paraview.org

ParaView



Export in EPX via ECRI

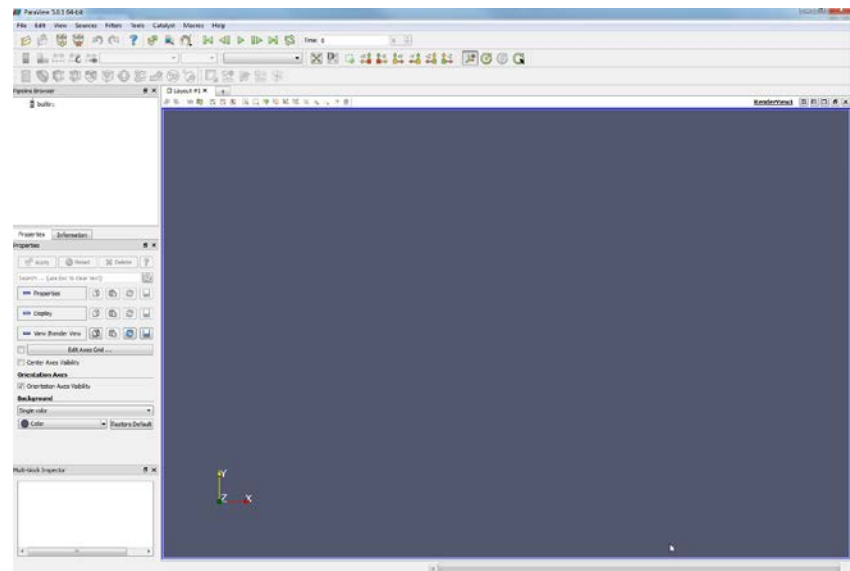
- ECRI FICH PVTk TFRE 5.E-5

GROUP AUTO

VARI DEPL VITE ACCE FEXT FINT ECRO

- vtu files were generated per block and per output step
- pvd file combines them → could be read in ParaView

→ Let's play with it!



Materials

- Europlexus manual

8 GROUP C1—MATERIALS

- **MATE** command

```
MATE GAZP RO 5.9485 GAMMA 1.4 CU 716.75  
      PINI 0.5E6 PREF 1.E5  
      LECT expl TERM  
GAZP RO 1.1897 GAMMA 1.4 CU 716.75  
      PINI 1.E5 PREF 1.E5  
      LECT air TERM
```

Fluid

```
UM23 RO 7800. YOUNG 1.6E11 NU 0.333 ELAS 1.05E8  
      TRAC 2 1.05E8 .656256E-3 1.6105E10 1.00066  
      LECT stru TERM
```

Structural

```
CLUF INFI RO 1.189 PRES 1.e5 GAMA 1.4 LECT abso TERM
```

Absorbing

Materials

NUMBER OF ELEMENTS TYPE	156
NUMBER OF MATERIALS TYPE	123

Material-Element combinations (from manual)

AVAILABLE MATERIALS FOR EACH ELEMENT

NO. | ELEMENT | AVAILABLE MATERIALS

1	COQU	LINE PARF ISOT TETA DYNA ORTH
2	TRIA	LINE PARF ISOT TETA POST FLUI CAVI GAZP NAH2 BETO
		DRUC DYNA EAU LIQU SOUR MULT FLFA STGN ADCR VM23
		PUFF ORTH JWLS CHOC ADCJ RSEA CRIT BUBB JWLS ZALM
		LMC2 LEM1 BOIS VMJC VMZA VMLP VMLU DRPR VMSF DPSF
		CAMC CLAY SGMP ENGR VPJC
3	BARR	LINE PARF ISOT DYNA
4	PONC	LINE PARF ISOT
5	MEMB	LINE
6	CUBB	LINE PARF ISOT DPDC
7	CL2D	IMPE CLVF IMPV
8	CAR1	LINE PARF ISOT TETA POST FLUI CAVI GAZP NAH2 BETO
		DRUC DYNA EAU LIQU SOUR MULT FLFA STGN ADCR VM23
		DONE PUFF ORTH JWLS CHOC ADCJ RSEA CRIT BUBB JWLS
		ZALM LMC2 LEM1 BOIS VMJC VMZA VMLP VMLU DRPR VMSF
		DPSF CAMC CLAY SGMP VPJC
9	CAR4	LINE PARF ISOT TETA POST FLUI CAVI GAZP BETO DRUC
		DYNA EAU MULT FLFA STGN VM23 DONE PUFF ORTH GLAS
		CHOC ADCJ CRIT BUBB LSGL ZALM LMC2 LEM1 BOIS VMJC
		VMZA VMLP VMLU DRPR VMSF DPSF CAMC CLAY HYPE ENGR
		VPJC
10	COQC	LINE PARF ISOT ORTH
11	CUBE	LINE ISOT TETA FLUI GAZP NAH2 DRUC DESM DYNA BLMT
		EAU LIQU SOUR MULT FLFA STGN ADCR VM23 PUFF ORTH
		ORTE GLAS ORTS JWLS CHOC ADCJ RSEA ORPE CRIT BUBB
		LSGL JWLS ZALM LMC2 LEM1 BOIS VMJC VMZA VMLP VMLU
		DRPR VMSF MAZA DPSF CAMC CLAY FOAM HYPE PBED TVMC
		SLZA CRTM SGMP EOBT DADC DPDC BDEM VPJC ORTP
12	COQ3	LINE ISOT TETA DYNA MCOU CHAN
13	CUB6	LINE ISOT TETA FLUI GAZP DRUC DYNA BLMT MULT STGN
		VM23 PUFF ORTH ORTE GLAS ORTS CHOC ORPE CRIT LSGL
		BOIS VMJC DRPR VMSF DPSF CAMC CLAY FOAM HYPE SLZA
		CRTM VPJC
14	COQ4	LINE ISOT TETA DYNA MCOU CHAN

AVAILABLE ELEMENTS FOR EACH MATERIAL

E AFTER MATERIAL INDICATES ERODIBLE.

NO. | MATERIAL | AVAILABLE ELEMENTS

1	LINE E	COQU TRIA BARR PONC MEMB CUBB CAR1 CAR4 COQC CUBE
		COQ3 CUB6 COQ4 POUT BR3D PR6 TETR TUYA PRIS PMAT
		CUB8 QPPS CMC3 T3GS BILL DST3 DKT3 SHB8 XCUB XCAR
		PROT SPHC Q4G4 MS24 S24 MS38 S38 Q4GR Q4GS INT4
		INT6 INT8 ASHB COQ2 Q4MC T3MC P3ZT TYVF
2	PARF E	COQU TRIA BARR PONC CUBB CAR1 CAR4 COQC BR3D XCUB
		XCAR
3	HILL	PR6 TETR PRIS CUB8 DST3 Q4GR Q4GS Q4MC T3MC P3ZT
4	ISOT E	COQU TRIA BARR PONC CUBB CAR1 CAR4 COQC CUBE COQ3
		CUB6 COQ4 POUT BR3D PR6 TETR TUYA PRIS CUB8 QPPS
		CMC3 T3GS BILL DST3 DKT3 SHB8 XCUB XCAR PROT SPHC
		Q4G4 Q4GR Q4GS ASHB COQ2 Q4MC T3MC P3ZT TYVF
5	TETA E	COQU TRIA CAR1 CAR4 CUBE COQ3 CUB6 COQ4 PR6 TETR
		PRIS CUB8 T3GS DST3 Q4GR Q4GS COQ2
6	POST	TRIA CAR1 CAR4
7	FLUI	TRIA CAR1 CAR4 CUBE CUB6 PR6 TETR TUBE TUYA BIFU
		CAVI PRIS CUB8 QAX1 TUBM PFEM TUYM T3VF Q4VF CUVF
		PRVF TEVF PYVF TUVF TYVF BIVF CAVF
8	CAVI	TRIA CAR1 CAR4
9	GAZP	TRIA CAR1 CAR4 CUBE CUB6 PR6 TETR TUBE TUYA BIFU
		CAVI PRIS CUB8 QAX1 TUBM TUYM T3VF Q4VF CUVF PRVF
		TEVF PYVF TUVF TYVF BIVF CAVF
10	NAH2	TRIA CAR1 CUBE TETR TUBE TUYA BIFU CAVI PRIS TUBM
		TUYM
11	BETO E	TRIA CAR1 CAR4 PR6 CMC3
12	DRUC E	TRIA CAR1 CAR4 CUBE CUB6 PR6 TETR PRIS CUB8
14	IFS	
15	DESM	CUBE PR6 TETR PRIS CUB8
16	IMPE	CL2D CL3D CL1D CL3T CLTU CL23 CL2S CL3S CL32 CL33
		CL22 CL3I CLD3 CLD6 CL3Q CL92 CL93
18	ODMS	CUB8 XCUB

Materials

NUMBER OF ELEMENTS TYPE	156
NUMBER OF MATERIALS TYPE	123

number	name	ref	law of behaviour
74	ABSE		
21	CLVF	7.8.33	Boundary conditions for finite volumes
109	DADC	7.6.16	Dynamic Anisotropic Damage Concrete
111	DPDC	7.6.17	dynamic plastic damage concrete
87	DPSF	7.6.41	Drucker Prager with softening and viscoplastic regularization
83	DRPR	7.6.50	Drucker Prager Ispra model
12	DRUC	7.6.6	Drucker-Prager
19	DYNA	7.6.7	dynamic Von Mises isotropic rate-dependent
17	FANT	7.6.29	phantom: ignore the associated elements
9	GAZP	7.7.4	perfect gas
118	GGAS	7.7.1	generic ideal gas material
116	GLIN	7.6.3	generic linear material
117	GPLA	7.6.4	generic plastic material
48	GVDW	7.7.27	Van Der Waals gas
40	GZPV	7.7.23	perfect gas for Van Leer
95	HYPE	7.6.53	hyperelastic material (Model of Mooney-Rivlin, Hart-Smith and Ogden)
4	ISOT	7.6.7	isotropic Von Mises
108	JCLM	7.6.64	Johnson-Cook with Damage Lemaitre-Chaboche for SPHC
50	JWL	7.7.20	explosion (Jones-Wilkins-Lee model)
66	JWLS	7.7.28	Explosion (Jones-Wilkins-Lee for solids)
72	LEM1	7.6.9	Von Mises isotropic coupled with damage (type Lemaitre)
1	LINE	7.6.1	linear elasticity
23	LIQU	7.7.14	incompressible (or quasi-) fluid
70	LMC2	7.6.11	Von Mises isotropic coupled with damage (Lemaitre) with strain-rate sensitivity
26	MASS	7.6.28	mass of a material point
85	MAZA	7.6.15	Mazars-linear elastic law with damage
2	PARF	7.6.7	perfectly plastic Von Mises
125	RIGI	7.6.66	Rigid material (for rigid bodies)
99	SLZA	7.6.55	Steinberg-Lund-Zerilli-Armstrong
35	VM23	7.6.37	Von Mises elasto-plastic radial return
2/4/5/19	VMIS	7.6.7	Von Mises materials
76	VMJC	7.6.46	Johnson-Cook
78	VMLP	7.6.47	Ludwig-Prandtl
79	VMLU	7.6.48	Ludwik
84	VMSF	7.6.40	Von Mises with softening and viscoplastic regularization
77	VMZA	7.6.49	Zerilli-Armstrong
120	VPJC	7.6.65	visco-plastic Johnson-Cook
67	ZALM	7.6.10	Zerilli-Armstrong with damage Lemaitre-Chaboche

Materials (input)

```
"CDEM" "TINI" tini "PINI" pini <"PREF" pref>
... "KSIO" ksi0 "KO" k0
... "TMAX" tmax "R" rgas
... "NESP" nesp "ORDP" ordp "NLHS" nlhs
...
... "COMP1"
... "MMOL" mmol "H0" h0 "CREA" crea
... "CV0" cv0 "CV1" cv1 ... "CVordp" cvordp
... "YMAS" ymas
...
...
... "COMPnesp"
... "MMOL" mmol "H0" h0 "CREA" crea
... "CV0" cv0 "CV1" cv1 ... "CVordp" cvordp
... "YMAS" ymas
...
...
... <"KOF" kof>
... <"UCDS" ucds>
... <"DIRE" dire>
... <"TO " temp0>
... <"H " hcoef>
... <"GX " gx>
... <"GY " gy>
... <"GZ " gz>
...
... <"CFLA" 1 "SL " sl "PSL " psl "TSL " tsl
... "DL " dl "LE " le "PU " pu
... "XIG " xig "YIG " yig "ZIG " zig>
... <"CFLA" 2 "SL " sl "PSL " psl "TSL " tsl
... "DL " dl "LE " le "A " acoef "B " bcoef
... "RO " r0 "XIG " xig "YIG " yig "ZIG " zig>
/LECT/
```

```
"GAZP" "RO" rho "GAMMA" gamma "PINI" pini <"VISC" mu > ...
... < "CV" cv > < "PREF" pref > /LECTURE/
```

```
"JWLS" "RO" rho <"ROS" ros> "PINI" pini <"PREF" pref> ...
... "A" a "B" b "R1" r1 "R2" r2 ...
... "OMEG" omeg "EINT" eint <"BETA" beta > ...
... < "D" d "XDET" xdet "YDET" ydet <"ZDET" zdet> > ...
... <"EAFT" eaft "CONF" conf "CHAR" char "TSTA" tsta "TEND" tend> ...
... /LECTURE/
```

```
"VM23" "RO" rho "YOUN" young "NU" nu "ELAS" sige ...
<"FAIL" $[ VMIS ; PEPS ; PRES ; PEPR ]$ "LIMI" limit>
"TRAC" npts*(sig eps)
/LECTURE/
```

```
"VPJC" "RO" rho "YOUN" young "NU" nu "ELAS" elas
<"TOL" tol "MXIT" mxit>
"QR1" qr1 "CR1" cr1 "QR2" qr2 "CR2" cr2
"PDOT" pdot "C" c
"TQ" tq "CP" cp <"TR" tr> "TM" tm "M" m
"DC" dc "WC" wc
<"SOLU" solu>
/LECTURE/
```

```
"DPDC" "RO" rho "YOUN" young "NU" nu "FC" fc
"DAGC" dagg
<"GFT" gft "GFC" gfc "GFS" gfs>
<"PWR" pwr> <"PWR" pwr>
<"B" b> <"D" d>
<"OVEC" overc> <"OVET" overt> <"SRAT" srate>
<"R" r> <"X0" xo> <"W" w>
<"D1" d1> <"D2" d2> <"PMOD" pmod>
<"TXCA" txca "TXCT" txct "TXCL" txcl "TXCB" txcb>
<"FTR" ftr "FBCR" fbc "IICR" iicr "RJCR" rjcr>
<"TXEA" txea "TXET" txet "TXEL" txel "TXEB" txeb>
<"NC" nc "NOC" noc> <"NT" nt "NOT" not>
<"REPW" repow> <"RECO" recov>
<"PRED" pred> <"COPP" copp> <"EXCT" excent>
<"VERS" vers>
<"EFVI">
<"EROD" <"ENDT" endt> <"ENDC" endc> <"DVOL" dvol>
/LECTURE/
```

```
"HYPE"
"TYPE" 4
"RO" rho
"C01" c1
"C02" c2
"C03" c3
"C04" c4
"C05" c5
"C06" c6
"C07" c7
"C08" c8
"BULK" K
/LECTURE/
```



Materials (types)

- **Structural**
 - Brittle (DPDC)
 - Ductile (VM23, VPJC, VMIS ISOT)
 - Hyperelastic (HYPE)
- **Fluid**
 - Perfect gas (GAZP)
 - Detonation in gas mixture (GAZD, CDEM)
 - Water (FLUI, ADCR, SGMP)
 - EOS Jones-Wilkins-Lee (JWLS)
- **Impedances** (CLVF, IMPE)
 - Absorbing boundary condition for fluid mesh (ABSO, INFI)
 - Loading boundary conditions (AIRB, PIMP)

Materials (capabilities)

- **Structural**
 - Large displacements/strains/rotations
 - Plasticity
 - Hardening / Softening
 - Strain rate effect
 - Erosion
 - hyperelasticity
- **Fluid**
 - Perfect gas
 - Detonation
 - Gas mixture
 - Combustion

Materials: internal variables (ECRO)

DPDC

ECR(1) : Pressure
 ECR(2) : Von Mises criterion
 ECR(3) : Equivalent plastic strain
 ECR(4) : Cube root of initial element volume (if version8)
 ECR(5) : Lode angle
 ECR(6) : Total variation of the isotropic hardening parameter
 ECR(7) : Volumetric strain
 ECR(8) : Plastic volumetric strain
 ECR(9) : Ductile damage parameter
 ECR(10): Brittle damage parameter
 ECR(11): Ductile damage threshold
 ECR(12): Brittle damage threshold
 ECR(13): Current damage
 ECR(14): Initial damage threshold in compression
 ECR(15): Initial damage threshold in tension
 ECR(16): filtered effective strain rate
 ECR(17-22): Components of elastoplastic stress tensor
 ECR(23-28): Components of viscoplastic stress tensor
 ECR(29): Effective strain rate
 ECR(30): Triaxiality
 ECR(31): Static ductile damage threshold
 ECR(32): Static brittle damage threshold

GAZP

ECR(1): pressure
 ECR(2): density
 ECR(3): velocity of sound
 ECR(4): maximum pressure ever experienced
 ECR(5): minimum pressure ever experienced
 ECR(6): dynamic pressure: ($P_{\text{dyn}} = \frac{1}{2}\rho v^2$)
 ECR(7): temperature (if c_V is not zero)
 ECR(8): total specific energy ($E = h + \frac{1}{2}v^2$)

VPJC

ECR(1) : SIGMAH. Hydrostatic pressure ($\frac{1}{3}\sigma_{kk}$)
 ECR(2) : PHI. Von Mises equivalent stress (σ_{eq})
 ECR(3) : P. Equivalent plastic strain (p)
 ECR(4) : PHITRIAL. Elastic trial equivalent (von Mises) stress
 ECR(5) : F. Yield function (which should be close to 0.0)
 ECR(6) : R. Total hardening of the material
 ECR(7) : DDLAMBDA. Change of the incremental plastic multiplier (from one time step to another)
 ECR(8) : DLAMBDA. Incremental plastic multiplier
 ECR(9) : NRITER. Number of iterations to obtain convergence
 ECR(10) : DLAMBDA / DT. Rate of plastic multiplier increment in time
 ECR(11) : D. Damage (D), i.e. fraction of voids with respect to the gross cross-sectional area
 ECR(12) : Failure indicator: 1.0 = Virgin Gauss Point, 0.0 = Failed Gauss Point
 ECR(13) : T. Absolute temperature (T)
 ECR(14) : WE. Cockcroft-Latham damage accumulation (W)
 ECR(15) : Sound speed
 ECR(16) : First principal stress (σ_1)
 ECR(17) : Second principal stress (σ_2)
 ECR(18) : Third principal stress (σ_3)

Elements

- Europlexus manual

3 SPATIAL DISCRETIZATION

```
EX03  
ECHO  
CONU WIN  
KFIL  
EROS 0.0  
TRID ALE  
DIME  
    NALE 1  
    NBLE 1  
TERM
```

Type of computation

```
GEOM Q4GS PART 1  
    CUUF hexa PART 2  
    PRUF penta PART 3  
    BR3D bars PART 4  
    CL3D absor PART 5  
TERM
```

Element definition

Elements

NUMBER OF ELEMENTS TYPE

156

The code contains various formulations, including:

- Finite Elements, which may be used for both structural and fluid parts of the model;
- Spectral Elements, which can be used for the discretization of continuum-like solid parts which behave linearly (e.g. small-strain wave propagation);
- Finite Volumes, which are suited for the fluid parts;
- SPH particles, often used for high-speed impact problems;
- Diffuse Elements.

1D CEA

num.	Name	Gibi	npt	dof	ngp	Remarks
22	TUBE	SEG2	2	1	1	fluid only (rigid tube)
23	TUYA	SEG2	2	7	2	tube coupled with FSI
24	CL1D	POI1	1	1	1	fluid boundary condition
25	BIFU	SUPE	1:9	7	1	bifurcation junction
26	CAVI	SUPE	1:9	1	1	cavity junction
31	CLTU	POI1	1	7	1	boundary condition with FSI
146	BREC	SEG2	2	7	1	pipeline rupture
147	TUVF	SEG2	2	1	1	rigid tube (1D vfcc)
148	TYVF	SEG2	2	7	1	tuvf coupled with FSI (1D vfcc)
149	BIVF	SEG2	2	1	1	bifurcation junction (1D vfcc)
150	CAVF	SEG2	1:9	1	1	cavity junction (1D vfcc)

Elements (2D)

CEA

num.	Name	Gibi	npt	dof	ngp	Remarks
1	COQU	SEG2	2	3	2	thin shell
2	TRIA	TRI3	3	2	1	triangle
3	BARR	SEG2	2	2	1	bar (membrane only)
4	PONC	POI1	1	2	1	circular bar(axisym.)
5	MEMB	SEG2	2	2	1	'virole' in membrane (axisym.)
7	CL2D	SEG2	2	2	1	boundary conditions
8	CAR1	QUA4	4	2	1	quadrangle with 1 Gauss pt.
9	CAR4	QUA4	4	2	4	quadrangle with 4 Gauss pt.
10	COQC	SEG2	2	3	1	thin shell with shear
15	FS2D	RAC2	4	2	1	F.S. coupling
28	PMAT	POI1	1	2	1	material point
45	TVL1	TRI3	3	2	1	Van Leer triangle
46	CVL1	QUA4	4	2	1	Van Leer quadrangle
114	BSHT	SEG2	2	2	-	bushing with translational dof
118	MAP2	---	3	2	-	Point on solid edge
121	MAP5	---	3	2	-	Point on 2D shell
124	INT4	---	4	2	2	2D interface element
131	T3VF	TRI3	3	2	1	triangle finite volume
132	Q4VF	QUA4	4	2	1	quadrangle finite volume

JRC

num.	Name	Gibi	npt	dof	ngp	Remarks
38	Q92	QUA8	9	2	4	9-node quadrilateral
39	Q93	QUA8	9	2	9	9-node quadrilateral
42	CL23	SEG3	3	2	2	3-node b.c.s
43	ED01	SEG2	2	3	10	beam/conical-shell
49	Q92A	QUA8	9	2	4	Q92 on symmetry axis
52	FLU1	QUA4	4	2	1	fluid quadrilateral
54	PFEM	POI1	1	2	1	particle finite element
56	ED41	SEG2	4	3	10	old version of ED01
58	ADQ4	QUA4	4	2	1	advection-diffusion quadrilateral
64	FL23	TRI3	3	2	1	fluid triangle
65	FL24	QUA4	4	2	1	fluid quadrilateral
70	CL22	SEG2	2	2	2	2-node b.c.s
71	Q41	QUA4	4	2	1	ALE structural quadrilateral
72	Q42	QUA4	4	2	4	ALE structural quadrilateral
73	Q41N	QUA4	4	2	1	ALE structural quadrilateral
74	Q42N	QUA4	4	2	4	ALE structural quadrilateral
75	Q41L	QUA4	4	2	1	Lagr. structural quadrilateral
76	Q42L	QUA4	4	2	4	Lagr. structural quadrilateral
77	Q95	QUA8	9	2	4	test version of Q92
97	MC23	TRI3	3	2	1	finite volume fluid triangle
98	MC24	QUA4	4	2	1	finite volume fluid quadrilateral
100	Q42G	QUA4	4	2	4	ALE structural quadrilateral
105	MS24	QUA4	4	2	1	spectral "macro" quadrilateral
106	S24	QUA4	4	2	1	spectral "micro" quadrilateral
109	FUN2	SEG2	2	2	1	cable element
140	DEBR	POI1	1	2	-	debris particle

Elements (3D)

CEA

num.	Name	Gibi	npt	dof	ngp	Remarks
6	CUBB	CUB8	8	3	8	brick based on B-bar method
11	CUBE	CUB8	8	3	1	brick with 1 Gauss pt
12	COQ3	TRI3	3	6	1	triangular thin shell
13	CUB6	CUB8	8	3	6	brick with 6 Gauss pt
14	COQ4	QUA4	4	6	4	quadrangular thin shell
16	FS3D	RAC3	8	3	1	F.S. connection (4-node face)
17	POUT	SEG2	2	6	2	beam
18	CL3D	QUA4	4	3	1	bound. cond. (4-node face)
19	BR3D	SEG2	2	3	1	spring or bar
20	PR6	PRI6	6	3	6	prism with 6 Gauss pt
21	TETR	TET4	4	3	1	tetrahedron with 1 Gauss pt
27	PRIS	PRI6	6	3	1	prism with 1 Gauss pt
28	PMAT	POI1	1	3	1	material point
29	CL3T	TRI3	3	3	1	bound. cond. (3-node face)
30	CUB8	CUB8	8	3	8	brick with 8 Gauss pt
32	APPU	POI1	1	6	1	support
33	MECA	SEG2	2	6	1	mechanism (articulated systems)
35	QPPS	QUA4	4	6	5	similar to Q4GR
41	TUBM	SUPE	1*	3*	1	connection 3D-1D
47	CMC3	TRI3	3	6	2	multilayer shell
48	FS3T	PRI6	6	3	1	F.S. connection (3-node face)
51	T3GS	TRI3	3	6	5	shell (Reissner-Mindlin)
79	BILL	POI1	1	3	1	particle element (NABOR and SPH)
80	ELDI	POI1	1	6	1	discrete element
81	CUVL	CUB8	8	3	1	Van Leer cube
82	PRVL	PRI6	6	3	1	Van Leer prism
83	DST3	TRI3	3	6	15	shell (Discrete Shear Triangle)
84	DKT3	TRI3	3	6	15	shell (Mindlin)
85	SHB8	CUB8	8	3	5	thick shell
89	SPHC	POI1	1	6	1	particle element (thick shell)
90	Q4G4	QUA4	4	6	4	shell (Batoz)
111	Q4GR	QUA4	4	6	5	idem Q4G4 (simplified : 1 pt)
112	Q4GS	QUA4	4	6	20	idem Q4G4 (simplified : 4 pts)
113	RL3D	SEG2	2	3	1	two-node spring
114	BSHT	—	2	3	—	bushing with translational dof
115	BSHR	—	2	6	—	bushing with trans. and rot. dof
116	TUYM	SUPE	1*	3*	1	connection 3D-1D
117	SH3D	—	3	6	—	node to shell connector
119	MAP3	—	4	3	—	point on a triangular facet
120	MAP4	—	5	3	—	point on a quadrangular facet

JRC

num.	Name	Gibi	npt	dof	ngp	Remarks
40	COQI	TRI3	3	6	15	triangular shell (small strain)
53	FLU3	CUB8	8	3	1	fluid brick
54	PFEM	POI1	1	3	1	particle finite element
57	ADC8	CUB8	8	3	1	advection-diffusion brick
62	CL32	QUA4	4	6	4	b.c.s for CQD4
63	CL33	QUA9	9	6	9	b.c.s for CQD9
66	FL34	TET4	4	3	1	fluid tetrahedron
67	FL35	PYR5	5	3	1	fluid pyramid
68	FL36	PRI6	6	3	1	fluid prism
69	FL38	CUB8	8	3	1	fluid hexahedron
78	CL3I	TRI3	3	3	1	3-node b.c.s
91	CQD4	QUA4	4	6	20	degenerated shell (Hughes-Liu)
92	CQD9	QUA9	9	6	45	degenerated shell (Hughes-Liu)
93	CQD3	TRI3	3	6	15	degenerated shell (Hughes-Liu)
94	CQD6	TRI6	6	6	20	degenerated shell (Hughes-Liu)
95	CLD3	TRI3	3	6	3	b.c.s for CQD3
96	CLD6	TRI6	6	6	4	b.c.s for CQD6
99	CL3Q	QUA4	4	3	1	4-node b.c.s
101	MC34	TET4	4	3	1	finite volume tetrahedron
102	MC35	PYR5	5	3	1	finite volume pyramid
103	MC36	PRI6	6	3	1	finite volume prism
104	MC38	CUB8	8	3	1	finite volume hexahedron
107	MS38	CUB8	8	3	1	spectral "macro" quadrilateral
108	S38	CUB8	8	3	1	spectral "micro" quadrilateral
110	FUN3	SEG2	2	3	1	cable element
140	DEBR	POI1	1	3	—	debris particle
144	C272	CU27	27	3	8	27-node cube
145	C273	CU27	27	3	27	27-node cube
151	CL92	QUA9	9	3	4	9-node (3D) b.c. for C272
152	CL93	QUA9	9	3	9	9-node (3D) b.c. for C273
155	C81L	CUB8	8	3	1	8-node cube
156	C82L	CUB8	8	3	8	8-node cube

Elements (3D)

- **Structural**
 - Beam like elements (BR3D)
 - Shell elements (Q4GS, T3GS)
 - Solid elements (CUBE, CUB8, PRIS, TETR)
 - Material points (PMAT, DEBR)
- **Fluid**
 - Finite volumes (CUVF, PRVF, TEVF, PYVF)
 - Finite elements (FL38, FL36, FL34)
- **Boundary condition**
 - Shell elements (CL3D, CL3T)

Storage of results

- Europlexus manual

12 GROUP G—PRINTOUT AND STORAGE OF RESULTS

ECRI DEPL ECROU VITE TFRE 0.5E-3
NOPO NOEL
ELEM LECT p1 TERM

Listing file

FICH ALICE TEMPS TFREQ 1.0e-4
POINT lect p_capt term
ELEM lect p1 p2 p3 term

Alice temp file

FICH PUTK TFRE 0.1e-3!FREQ 0 TFRE 0.00
GROU AUTO
VARI UCVI ECRO

Paraview file

Storage of results

- **Nodal**
 - DEPL (displacement)
 - VITE (velocity)
 - ACCE (acceleration)
 - FEXT (external forces)
 - FLIA (link forces)
- **Elemental**
 - CONT (stress)
 - EPST (strain)
 - ECRO
 - RISK
 - FAIL
 - VCVI (fluid only)
 - ENER

Storage of results

- **Listing file (Ascii)**
 - Element/ node printouts with a defined output frequency
 - not very efficient
- **Alice file (binary)**
 - Europlexus storage type
 - Can be split
 - Is becoming very large but contains all the desired data
- **Alice temp file (binary)**
 - Europlexus storage type
 - Can be split
 - Very efficient but the user has to predefine the desired data
- **Paraview file (binary/ascii)**
 - Paraview storage type (pvtk)
 - Very efficient for visualizing purposes

Post-processing with EPX

15 GROUP ED—POST-TREATMENT BY EUROPLEXUS

```
CALC TINI 0.0 TFIN 200.E-3
```

```
SUIT
```

```
EXPL03AP
```

```
ECHO
```

```
RESU ALICE TEMPS GARD PSCR
```

```
SORT GRAP
```

```
AXTE 1.0 'Time [s]'
```

```
COUR 1 'Pressure' ECR0 COMP 1 ELEM LECT e1 TERM
```

```
COUR 2 'Pressure' ECR0 COMP 1 ELEM LECT e2 TERM
```

```
TRAC 1 2 AXES 1.0 'Pressure' YZER
```

```
COLO bleu roug
```

```
THIC 0.8 0.8
```

```
LIST 1 2 AXES 1.0 'Pressure' YZER
```

```
COUR 11 'UCVI' UCVI NORM ELEM LECT e1 TERM
```

```
COUR 12 'UCVI' UCVI NORM ELEM LECT e2 TERM
```

```
TRAC 11 12 AXES 1.0 'Velocity' YZER
```

```
COLO bleu roug
```

```
THIC 0.8 0.8
```

```
LIST 11 12 AXES 1.0 'Velocity' YZER
```

```
FIN
```

Read alice file

Create output curves

Print curves on a ps file

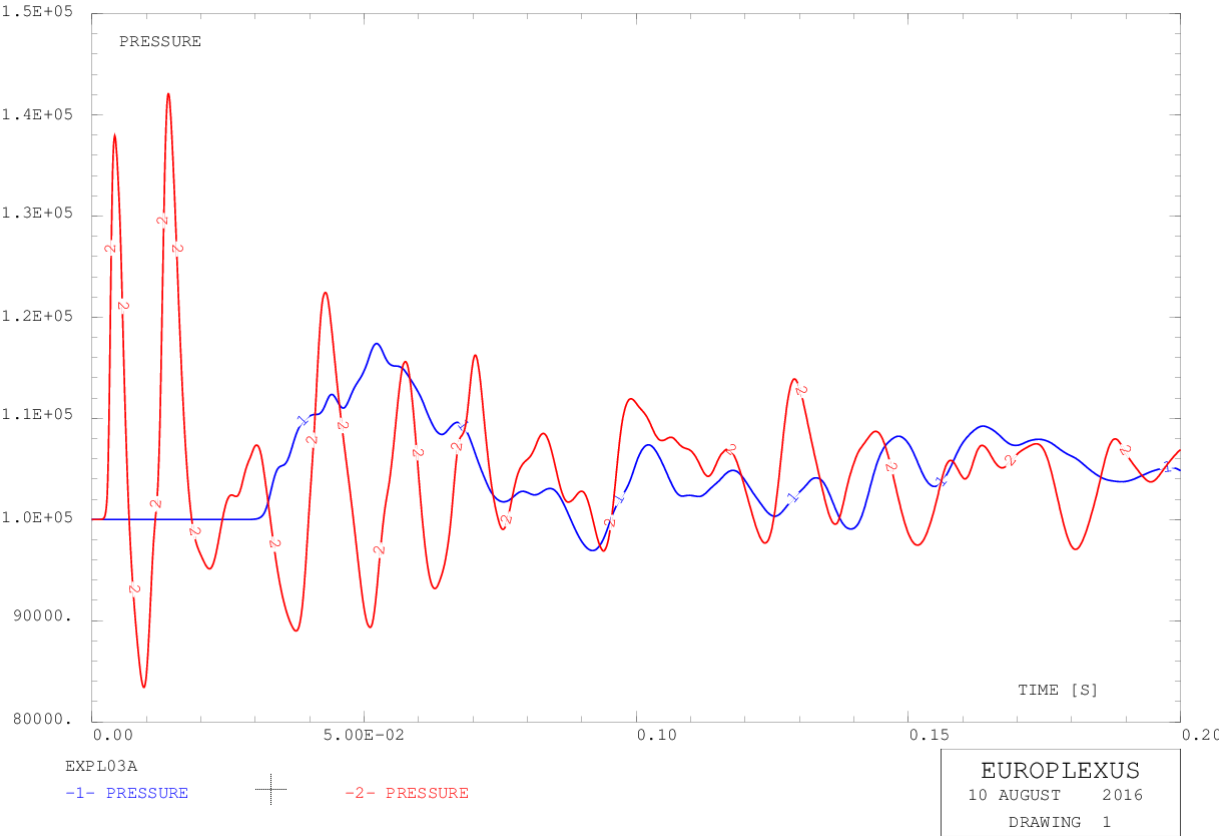
Print curves on a punch file

Post-processing with EPX - Curves

- Curves
- Nodal/ elemental quantities
- Zone quantities
- Curves as a function of space (SCOURBE)
- Read a curve (RCOURBE)
- Other special type of curves (PCOURBE, DCOURBE etc.)
- Curves manipulation
 - Integration
 - Summation
 - Max, mean, min
- **Postscript file creation (TRACE)**
 - Axis definition
 - Color, thickness etc.
 - Max, mean, min
- **Punch file creation (LIST)**

Post-processing with EPX

Post-script file example



Punch file example

VALUES	11	COMPONENTS	1
* Time [s]		Pressure1	
* T		Pressure	
0.000000E+00		1.000000E+05	
1.201111E-04		1.000000E+05	
2.161999E-04		1.000000E+05	
3.122888E-04		1.000000E+05	
4.083777E-04		1.000000E+05	
5.044665E-04		1.000000E+05	
6.005554E-04		1.000000E+05	
7.206665E-04		1.000000E+05	
8.167553E-04		1.000000E+05	
9.128442E-04		1.000000E+05	
1.008933E-03		1.000000E+05	
VALUES	11	COMPONENTS	1
* Time [s]		Pressure2	
* T		Pressure	
0.000000E+00		1.000000E+05	
1.201111E-04		1.000000E+05	
2.161999E-04		1.000000E+05	
3.122888E-04		1.000000E+05	
4.083777E-04		1.000000E+05	
5.044665E-04		1.000000E+05	
6.005554E-04		1.000000E+05	
7.206665E-04		1.000000E+05	
8.167553E-04		1.000000E+05	
9.128442E-04		1.000000E+05	
VALUES	11	COMPONENTS	1
* Time [s]		Velocity	
* T		UCVI	
0.000000E+00		0.000000E+00	
1.201111E-04		0.000000E+00	
2.161999E-04		0.000000E+00	
3.122888E-04		0.000000E+00	
4.083777E-04		0.000000E+00	
5.044665E-04		0.000000E+00	
6.005554E-04		0.000000E+00	
7.206665E-04		0.000000E+00	
8.167553E-04		0.000000E+00	
9.128442E-04		0.000000E+00	
1.008933E-03		0.000000E+00	

Post-processing with EPX

- REGIONS (11.9)

- Group of Nodes or Elements
- Monitor the quantities of a group
- Energy balance or average values
- Listing, punch , alics files
- Be aware of the available quantities
- The group has to be defined in advance
- Zones solution

```
"REGION" ( 'nom region'
$[ "RMAS" ; "VOLU" ; "BARY" ; "DIMX" ; "DIMN" ;
"DMOY" ; "VEMX" ; "VEMN" ; "VMOY" ; "ACMX" ;
"ACMN" ; "AMOY" ; "IMPU" ; "ECIN" ; "WINT" ;
"WEUT" ; "PDV" ; "WINJ" ; "RESU" ; "IRES" ;
"ECRG" ; "ECRM" ; "EMAS" ; "FLIR" ; "RISK" ;
"EROD" ; "ENDO" ; "CLAS" ; "EPSM" ; "TOUT" ]$
< "DIRX" rx "DIRY" ry "DIRZ" rz >
|[ /LECTURE/ ;
"POIN" /LECTURE/ ]| )
```

REGION : PLATE

INTERNAL ENERGY (WINT)	=	3.01728E+03
TRAVAIL DES PRESSIONS (PDU)	=	-1.72803E+01
INTRODUCED ENERGIE	=	0.00000E+00
KINETIC ENERGIE (ECIN)	=	4.34258E+02
KINETIC ENERGIE (COMP.)	=	4.94800E-03 9.31900E-03 4.34243E+02
MASS (COMPOSANTES)	=	1.78668E+01 1.78668E+01 1.78668E+01
MASS (SUM OF ELEMENTS)	=	1.78668E+01
DISPLAC. MAXIMUM (COMP.)	=	7.15988E-04 7.35179E-04 1.28955E-01 1.28955E-01
DISPLAC. MINIMUM (COMP.)	=	1.70512E-09 2.75385E-09 5.91600E-02 5.91612E-02
DISPLAC. AVERAGE (COMP.)	=	-6.43949E-08 -2.62663E-06 1.04001E-01
VELOCITY MAXIMUM (COMP.)	=	1.12027E-01 1.07962E-01 9.73004E+00 9.73028E+00
VELOCITY MINIMUM (COMP.)	=	1.79196E-06 1.14166E-06 4.70000E+00 4.70010E+00
VELOCITY AVERAGE (COMP.)	=	6.47717E-06 -9.60593E-04 6.83815E+00
ACCELER. MAXIMUM (COMP.)	=	3.23449E+02 6.26475E+02 3.88293E+03 3.88312E+03
ACCELER. MINIMUM (COMP.)	=	2.58365E-03 4.61048E-03 1.78445E-02 4.96671E+00
ACCELER. AVERAGE (COMP.)	=	1.64996E+01 3.29587E+01 2.24849E+02
IMPULSION (COMPOSANTES)	=	1.15726E-04 -1.71625E-02 1.22176E+02
BARYCENTRE (COMPOSANTES)	=	1.25000E-01 1.44997E-01 9.90000E-02
VOLUME	=	6.38204E-03
ENERGIE MECANIQUE (WEXT)	=	0.00000E+00
SUM (FEXT.DT) (COMPOS.)	=	0.00000E+00 0.00000E+00 0.00000E+00
RESULTANTE (COMPOSANTES)	=	0.00000E+00 0.00000E+00 0.00000E+00
SOMME(REGION) ECR (COMP.)	=	-3.72750E+00 6.38530E+11 7.21494E+00 0.00000E+00 4.11774E+04
R.F.LIAI/LINK (COMPOSANTES)	=	0.00000E+00 5.50080E+12 0.00000E+00 0.00000E+00 0.00000E+00
NBRE ELTS CASSES	=	0
NBRE ELTS PARTIELLEMENT CASSES	=	0
ELTS REGION / = ELTS CLASSES :AUCUNE	=	0.00000E+00
TOTAL RISK (EARDRUM)	=	0.00000E+00
TOTAL RISK (DEATH)	=	0.00000E+00
EPST (COMP.)	=	7.15564E-05 1.57330E-04 -2.28067E-04 1.04649E-07 4.40190E-07
		5.68338E-08 0.00000E+00 0.00000E+00 0.00000E+00 0.00000E+00
		0.00000E+00 0.00000E+00

Interactive commands with EPX

- EPX manual
- 15 GROUP O—INTERACTIVE COMMANDS
- CONV WIN (only for windows)

IMPA01
ECHO
CONV WIN
KFIL

```

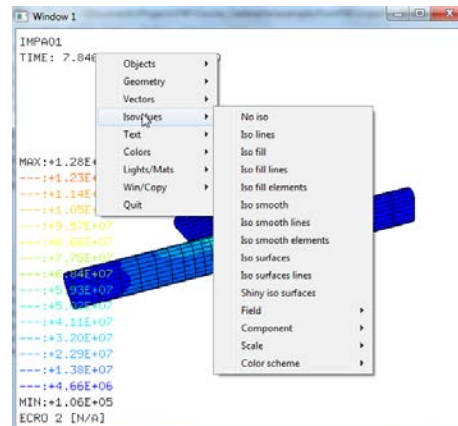
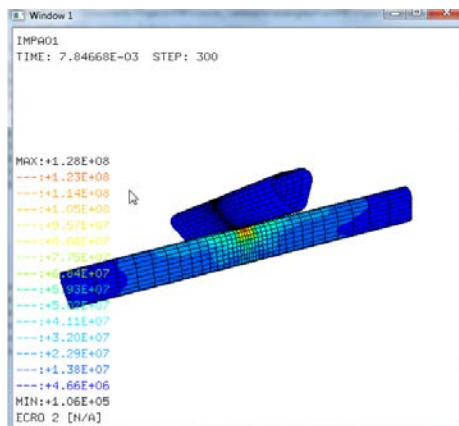
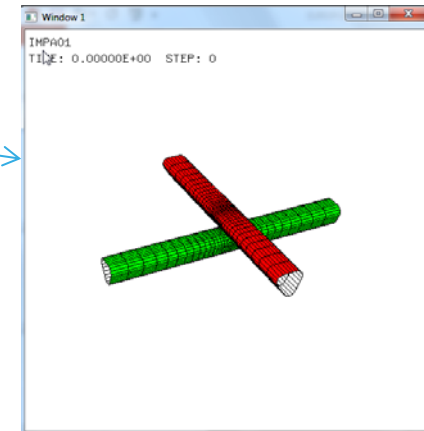
-> END OF INITIALIZATIONS ** TCPU = 0.12 SEC.

COMMAND ?
trac rend
64>trac rend
OPTION ?

** ATTENTION 1 IN THE ROUTINE RENDER_NGRPS ** THE MESSAGE IS THE FOLLOWING ONE :
NODE GROUPS ARENT DISJOINT

COMMAND ?
Freq 300 go
65>freq 300 go
PREVIOUS FREQUENCY (BY STEP) OF VISUALISATION: 1
CHOOSE NEW FREQUENCY:
NEW FREQUENCY (BY STEP) OF VISUALISATION: 300
COMMAND ?
> END OF TIME STEP NO : 0 * T = 0.00000E+00 * TCPU = 2.23 SEC.
> END OF TIME STEP NO : 39 * T = 1.02121E-03 * TCPU = 8.95 SEC.
> END OF TIME STEP NO : 77 * T = 2.01588E-03 * TCPU = 13.31 SEC.
> END OF TIME STEP NO : 115 * T = 3.01008E-03 * TCPU = 15.24 SEC.
> END OF TIME STEP NO : 153 * T = 4.00372E-03 * TCPU = 17.24 SEC.
> END OF TIME STEP NO : 192 * T = 5.02341E-03 * TCPU = 19.28 SEC.
> END OF TIME STEP NO : 230 * T = 6.01708E-03 * TCPU = 21.34 SEC.
> END OF TIME STEP NO : 268 * T = 7.01070E-03 * TCPU = 23.51 SEC.

COMMAND ?
trac obje lect PART 1 PART 2 term rend
66>trac obje lect PART 1 PART 2 term rend
OPTION ?
OPTION ?
COMMAND ?
stop_
    
```



Interactive commands with EPX

15 GROUP O—INTERACTIVE COMMANDS

- "TRAC REND" or "TRAC OBJECT ... TERM REND"
- "FREQ .." or "TFREQ .."
- "TIME"
- "INFO"
- "HPIN", "NOHP"
- "GO"
- "STOP"

Boundary conditions

8 GROUP D—LINKS

- Coupled Lagrange multipliers (implicit)**

- Cholesky
- Iterative solvers (Pardiso etc)
- BLOQ, FLSR, PINBALLS, CONT SPLA

```
LINK COUP SPLT NONE
      SOLU PARDISO
      BLOQ 123 LECT Fix3 TERM
Armat: beton lecture wal TERM
      ferr lecture bars term
      PINB BODY LECT FOA TERM
      BODY LECT WALC TERM
      BODY LECT WALF1 WALF2 TERM
CONT SPLA NX 0 NY 1 NZ 0 LECT symy TERM
CONT SPLA NX 0 NY 0 NZ 1 LECT symz TERM
```

- Decoupled (explicit)**

- Imposed via ad-hoc direct methods
- BLOQ, FLSW, PINBALLS (penalty method)

```
LINK DECO
      PINB PENA SFAC 1.0
      BODY FROT MUST 0.15 MUDY 0.1 GAMM 0 !MLEV 5
      DMIN 0.0008
      LECT lframeb TERM
      BODY FROT MUST 0.15 MUDY 0.1 GAMM 0 !MLEV 5
      DMIN 0.0008
      LECT uframeb TERM
      FLSW STRU LECT scoup TERM
      FLUI LECT fcoup TERM
      R 0.0088 ! 0.87*H_FLUID = 0.87*0.01 = 0.0087
      HGRI 0.0052 ! Slightly bigger than the fully refined
      ! fluid element (for this case)
      ! h_f^ref = h_f^base / 2^(lmax-1) =
      ! 0.01 / 2^1 = 0.005
```

Finding the Lagrange Multipliers (2)

- The CD scheme for the velocity and constant Δt is:

$$\underline{v}^{n+3/2} = \underline{v}^{n+1/2} + \Delta t \cdot \underline{a}^{n+1}$$
- Substituting this into the constraint $\underline{Cv} = \underline{b}$ gives:

$$\underline{Cv}^{n+3/2} = \underline{Cv}^{n+1/2} + \Delta t \cdot \underline{Ca}^{n+1} = \underline{b}$$
- From this we obtain:

$$\underline{Ca} = \frac{1}{\Delta t} (\underline{b} - \underline{Cv}^{n+1/2}) = \frac{1}{\bar{\gamma}} (\underline{b} - \underline{Cv}^{n+1/2}) \quad \text{having posed: } \bar{\gamma} = \Delta t$$
- For a variable Δt in time, one has simply:

$$\bar{\gamma} = \frac{\Delta t^n + \Delta t^{n+1}}{2}$$

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Boundary conditions

- Blockages (8.5)
- Time-limited blockages (8.6)
- Symmetry conditions (8.8.6)
- Sliding (8.27, 8.48)

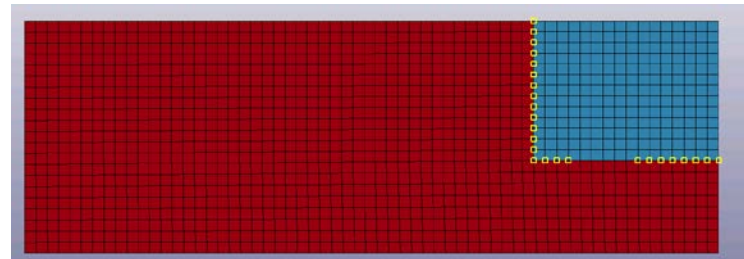
Fluid boundary conditions

- **Absorbing boundaries**
 - IMPE ABSO (FE)
 - CLVF ABSO, INFI (VFCC)
 - Applied via appropriate (CL) elements
- **Reflecting boundaries**
 - Every free face in 3D or edge in 2D (for FV)
 - FSR for FE

[Slides\videos\video2.avi](#)

[Slides\videos\GFM1.avi](#)

[Slides\videos\full21b2avi_01.avi](#)



Loading conditions

- **Loads**

10 GROUP F—LOADS

- Gravity
- Rotational
- Factorized

```
CHAR CONS GRAV 0.0 0.0 -9.81
LECT _DEBR TERM ! gravity acts only on debris particles
```

- **Impedances (7.8)**

- Load via CL elements
- Imposed pressure (PIMP)
- Blast load (AIRB)

```
IMPE PIMP RO 7850.0 PRES 43.1e6 PREF 1.0E5
LECT presur TERM
IMPE PIMP RO 7850 PRES 1.0 PREF 1.0E5 FONC 1
LECT preplat TERM
IMPE AIRB X 0.0 Y 0.0 Z -0.625
MASS 0.0402 TAUT
LECT pblast TERM
```

- **Initial conditions**

9 GROUP E—FUNCTIONS AND INITIAL CONDITIONS

- Initial velocity, stress (INIT)
- Harmonic function

```
INIT VITE 3 -12.0 LECT MES FOA MAS TERM
```

Contact models

- Sliding surfaces “GLIS” (8.27)

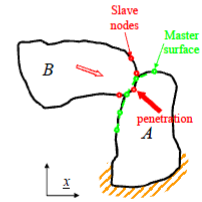
- coup, deco
- Less robust

```
GLIS 2 FROT MUST 0.15 MUDY 0.1 GAMM 0 ! Contact surface #1
PGAP 0.4E-3
MAIT LECT lframe TERM
PESC LECT plateN TERM
FROT MUST 0.15 MUDY 0.1 GAMM 0 ! Contact surface #2
PGAP 0.4E-3
MAIT LECT uframeb TERM
PESC LECT plateN TERM
```

Conventional contact-impact methods

- Slide-line (2D) or slide-surface (3D)
[Hallquist, Benson, mid-'80s]

- “Slave” nodes
- “Master” surfaces
- Contact detected as node-through-surface penetration



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- Pinball model “PINB” (8.48)

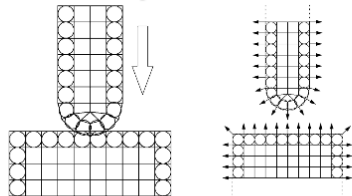
- coup, deco
- More robust for impact problems
- Efficient for complicated geometries

```
PINB BODY LECT FOA TERM
BODY LECT WALC TERM
BODY LECT WALF1 WALF2 TERM
BODY LECT SUPP TERM
EXCL PAIR 1 3
EXCL PAIR 1 4
EXCL PAIR 2 3
EXCL PAIR 2 4
```

The Basic Pinball Method

[Belytschko & Neal, 1991]

- Embed a sphere (pinball) in each element

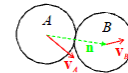


34

Contact Force Computation

- Constraints on velocities:

$$\mathbf{v}_A \cdot \hat{\mathbf{n}} - \mathbf{v}_B \cdot \hat{\mathbf{n}} \leq 0$$



- By expanding the velocities:

$$\left[\sum_{i=1}^{n_A} N_{Ai}(A) \mathbf{v}_{Ai} \right] \cdot \hat{\mathbf{n}} - \left[\sum_{i=1}^{n_B} N_{Bi}(B) \mathbf{v}_{Bi} \right] \cdot \hat{\mathbf{n}} \leq 0$$

- These inequalities are of the generic form:

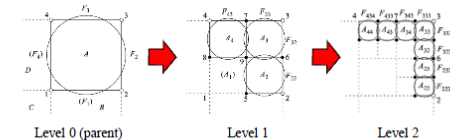
$$\mathbf{Cv} \leq \mathbf{b}$$

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The Hierarchic Pinball Method

[Belytschko & Yeh, 1993]

- Embed a **parent** pinball in each element
- Recursively generate descendent pinballs as long as contact holds

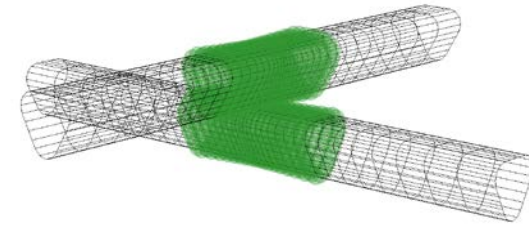


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Pinball model

- **Drop analysis**

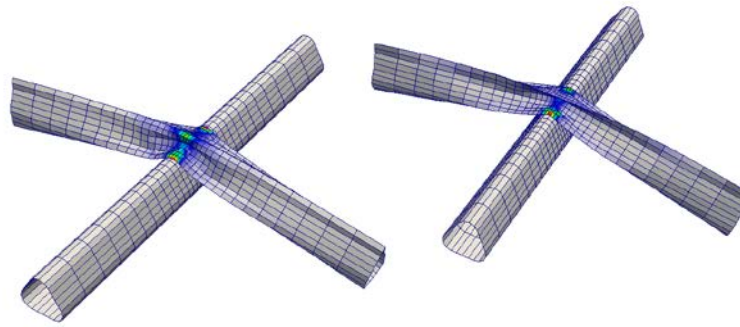
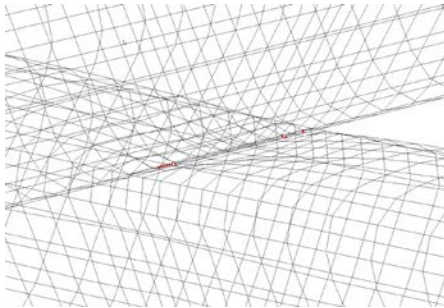
- 2 bodies (beams)
- One fixed the other with uniform velocity
- Same material different thickness



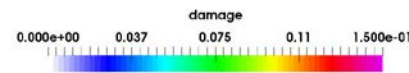
- **Pinball parameters**

- Hierarchy level for descendants (MLEV)
- Self-contact (SELF)
- Exclude pairs (EXCL)

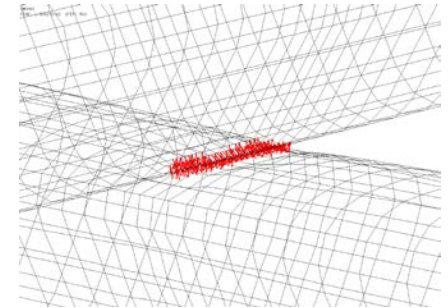
MLEV 0



Time: 0.250 sec



MLEV 2



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Pinball model

- Drop analysis

- Linear – Linear, uniform velocity

[Slides\videos\An1.avi](#)

```
23 MATE LINE RO 7800 YOUN 2.E11 NU 0.3
24      LECT PART 1 PART 2 TERM

33 LINK COUP SPLT NONE SOLV PARD
34      BLOQ 123 LECT fix1 fix2 TERM
35      PINB BODY LECT bod1 TERM
36      BODY LECT bod2 TERM
37 INIT VITE 2 30 LECT PART 2 TERM
```

- Plastic – Linear , uniform velocity (5 times slower)

[Slides\videos\An2.avi](#)

```
23 MATE LINE RO 7800 YOUN 2.E11 NU 0.3
24      LECT PART 1 TERM
25      VPJC RO 7850.0 YOUN 2.1E11 NU 0.33 ELAS 3.257E8 mxit 500
26      QR1 2.348E8 CR1 56.2 QR2 4.457E8 CR2 4.7
27      PDOT 5.E-4 C 1.E-2 TQ 0.9 CP 452.0
28      TM 1800.0 M 1.0 DC 1.0 WC 555.0E6
29      LECT PART 2 TERM
```

- Plastic – Plastic, rotational velocity

[Slides\videos\An5.avi](#)

```
33 LINK COUP SPLT NONE SOLV PARD
34      BLOQ 123 LECT fix1 fix2 TERM
35      BLOQ 123 LECT fix3 TERM
36      PINB BODY LECT bod1 TERM
37      BODY LECT bod2 TERM
38 CHAR CONS GRAV 0 9.80665 0 LECT tous TERM
39 INIT ROTA ORIG -20 1.0 -0.5764
40      VECT 0 0 1
41      OMEG 1.2
42      LECT PART 2 TERM
```

Impedances

Superposed elements (CL3D, CL3T)

- Shell: pressure
- Fluids: Absorbing boundaries

Material IMPE for loading

IMPE AIRB NODE LECT charge TERM MASS 1
LECT impe TERM

Fluid-Structure Interaction (FSI)

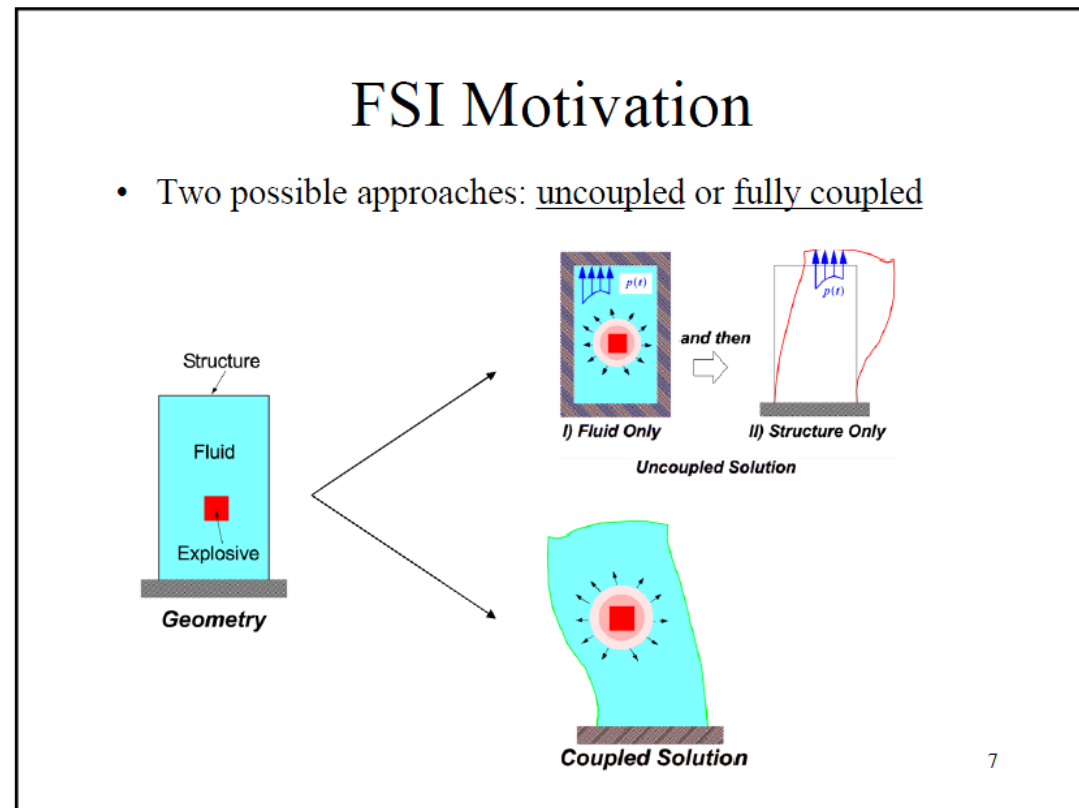
Fluid-structure interaction motivation

- **Fluid behavior**

- Reflections
- Channeling effect
- Shadowing effect

- **Structural behavior**

- Significant deformation
- Failure and element erosion
- Formation of flying debris



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Fluid-structure interaction

- **Fluid domain**

- Eulerian
- Compressible and inviscid
- Discretization: FE, NCFV, FVCC

- **Structural domain**

- Lagrangian
- Discretization: FE

- **FSI enforcement**

- Strong, constrains on the velocities (implicit)
- Weak, pressures force/ fluxes (explicit)

- **FSI discretization**

- Conforming mesh
- Non-conforming mesh (embedded)

Euler equations (2)

$$\frac{dM}{dt} = \frac{d}{dt} \int_{V(t)} \rho dV = \oint_{S(t)} \underbrace{\rho(\mathbf{w}-\mathbf{v}) \cdot \mathbf{n}}_{\text{transport}} dS \quad (\text{Mass})$$

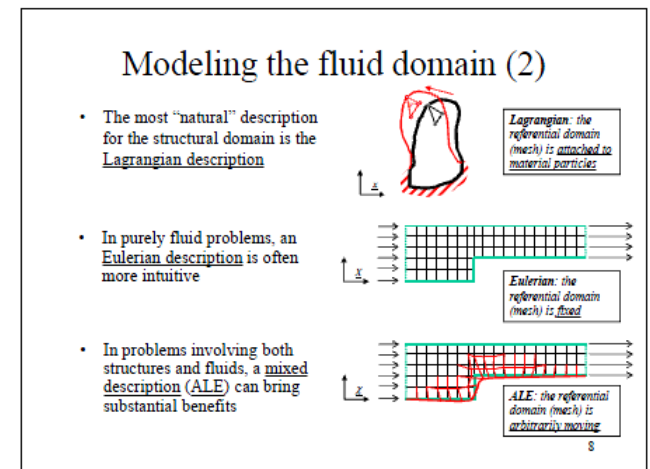
$$\frac{dQ}{dt} = \frac{d}{dt} \int_{V(t)} \rho \mathbf{v} dV = \oint_{S(t)} \underbrace{\rho \mathbf{v}(\mathbf{w}-\mathbf{v}) \cdot \mathbf{n}}_{\text{transport}} dS - \int_{V(t)} \underbrace{\nabla p}_{\text{pressure}} dV + \int_{V(t)} \underbrace{\rho \mathbf{g}}_{\text{body force}} dV \quad (\text{Momentum})$$

$$\frac{dE}{dt} = \frac{d}{dt} \int_{V(t)} \rho e dV = \oint_{S(t)} \underbrace{\rho e(\mathbf{w}-\mathbf{v}) \cdot \mathbf{n}}_{\text{transport}} dS - \oint_{S(t)} \underbrace{p \mathbf{v} \cdot \mathbf{n}}_{\text{pressure}} dS + \int_{V(t)} \underbrace{\rho \mathbf{g} \cdot \mathbf{v}}_{\text{body force}} dV \quad (\text{Energy})$$

M = mass of control volume ρ = fluid density e = total specific energy
 Q = momentum of control vol. p = pressure ∇ = gradient operator
 E = energy of control vol. \mathbf{g} = gravity \cdot = scalar product

Plus suitable equation of state: $p = p(\rho, T)$

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FSI discretization

Conforming mesh

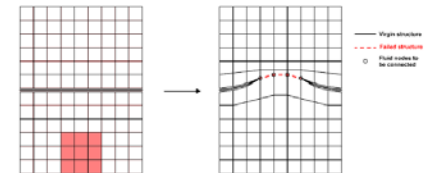
- Medium computational cost +
- Robust +
- Rezoning problems -
- No structural failure -
- High complexity in the preparation of the mesh -

Non-conforming mesh

- Treats cases with significant deformation (failure etc.) +
- Low complexity in the preparation of the mesh +
- Robust +
- High computational cost -
- Sensitive to the parameters of the influence domain -

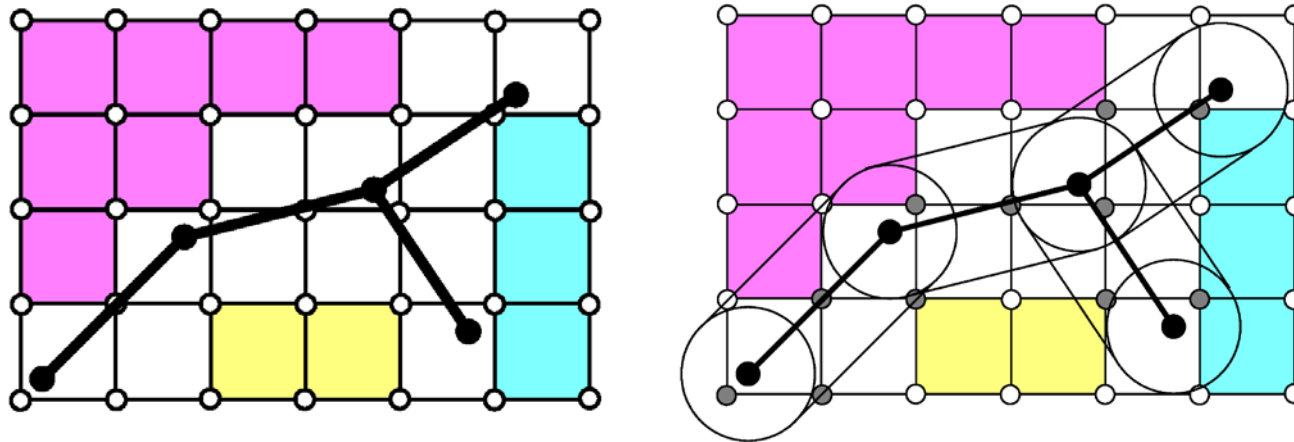
Issues in FSI with failure/fragmentation

- When structure fails and is eroded, the FSI models seen so far are no longer applicable, because:
 - Difficult to “sew” the fluid meshes on either side of the failing structure
 - Automatic fluid mesh rezoning algorithms fail because the master domain (structure) “disappears”
 - Possible rotating macro fragments create rezoning problems
 - Etc. etc.



FSI embedded model

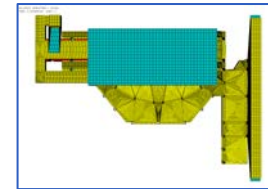
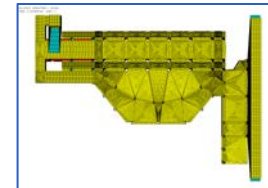
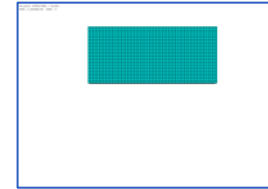
- Automatically build up an “influence domain” around the structure (a sphere at each S node)
- Identify (fast search) fluid nodes F currently located within the influence domain
- Impose suitable constraints on velocities : $\mathbf{v}_F \cdot \mathbf{n}_{S^*} - \mathbf{v}_{S^*} \cdot \mathbf{n}_{S^*} = 0$ (\mathbf{n}_{S^*} is the normal to the structure, S^* is closest structural point to fluid node)



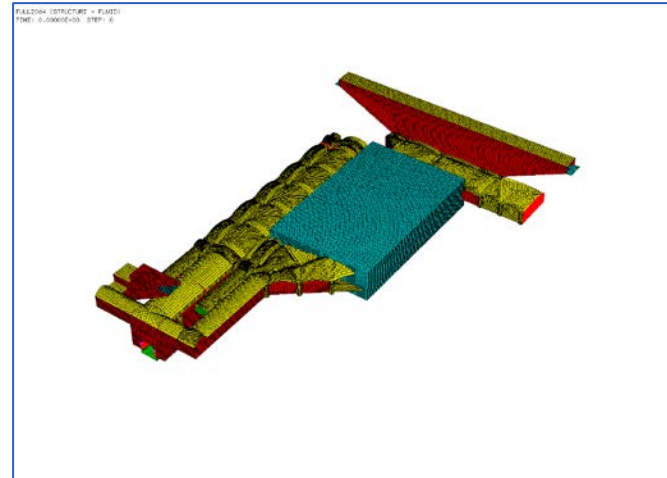
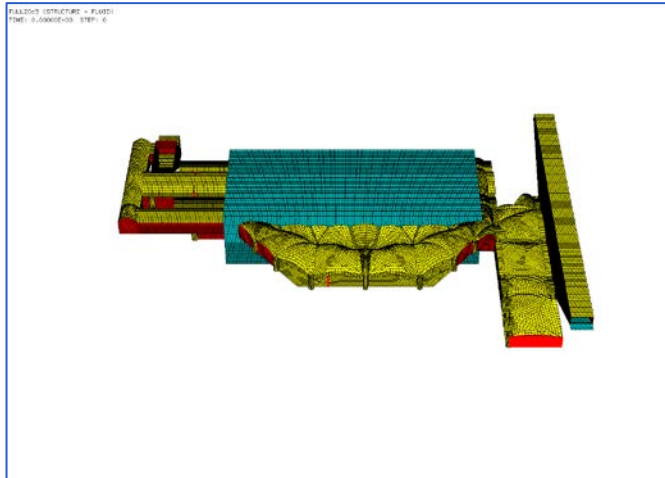
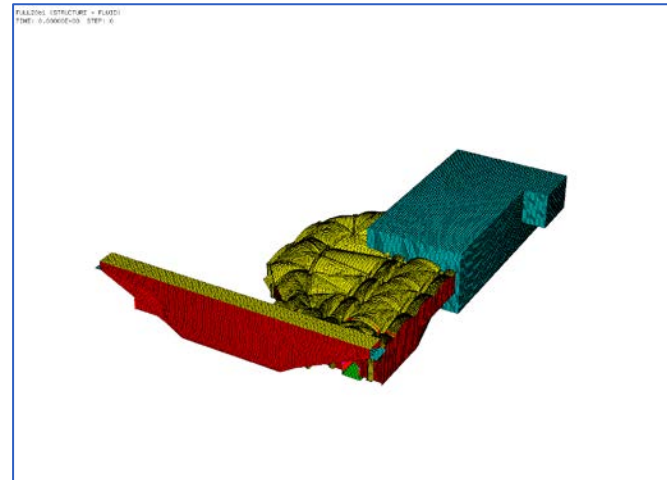
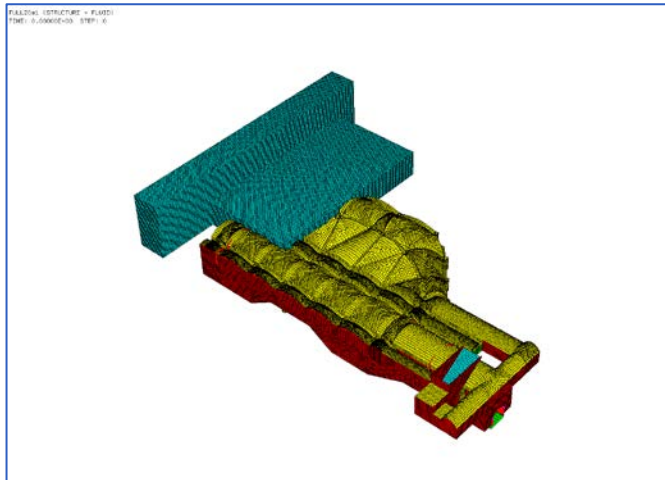
The grayed fluid nodes are coupled with the structure

FSI embedded model

- Fluid mesh regular parallelepiped grid
- Structural mesh
- The two meshes are simply superposed
- Absorbing boundaries on the envelope of the fluid mesh



FSI embedded model



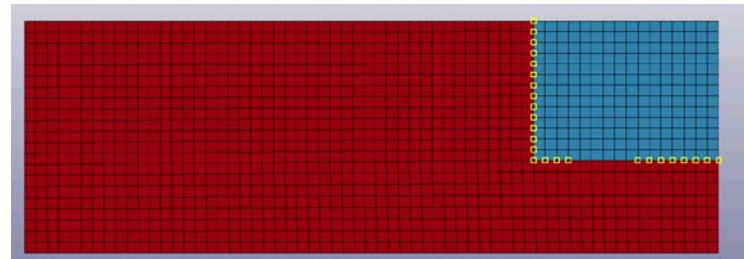
Fluid boundary conditions

- **Absorbing boundaries**
 - IMPE ABSO (FE)
 - CLVF ABSO, INFI (VFCC)
 - Applied via appropriate (CL) elements
- **Reflecting boundaries**
 - Every free face in 3D or edge in 2D (for FV)
 - FSR for FE

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[Slides\videos\GFM1.avi](#)

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FSI Approaches (embedded)

- **General**

- FE for the fluid mesh –
- Coupled –
- Robust +

- **Input**

- Finite El. 2D: FL24, FLUT
- Finite El. 3D: FL38, FLUT
- Abs. Bound. 3D: CL3Q, IMPE ABSI

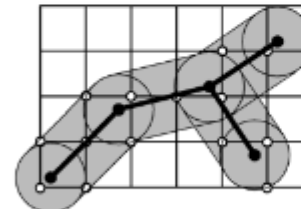
Strong embedded approach (FLSR)

- Detection via **influence domain** around structure
- **Fast** search of coupled fluid nodes
- Strong coupling with closest structure point

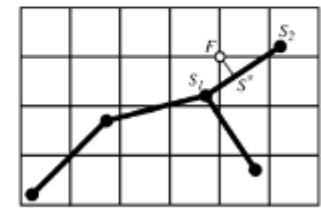
$$\mathbf{v}_p \cdot \mathbf{a}_2 = \mathbf{v}_p' \cdot \mathbf{a}_2 = (N_1 \mathbf{v}_{S_1} + N_2 \mathbf{v}_{S_2}) \cdot \mathbf{a}_2$$

or more simply :

$$\mathbf{v}_p = \mathbf{v}_p' = N_1 \mathbf{v}_{S_1} + N_2 \mathbf{v}_{S_2}$$



a) Influence domain (shaded) of the structure.



b) F-S coupling.

The FLSR embedded algorithm with a strong approach

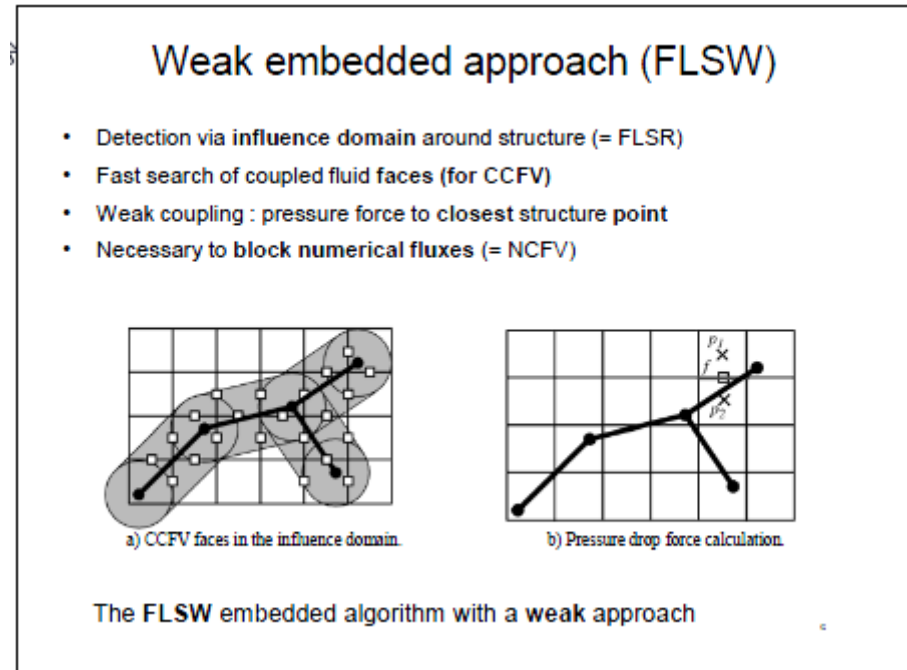
FSI Approaches (embedded)

- **General**

- FV for the fluid mesh +
- Decoupled +
- Robust +

- **Inputs**

- Finite Vol 1D: TUVF, GAZP
- Finite Vol 2D: Q4VF, GAZP
- Finite Vol 3D: CUVF, GAZP
- Abs. Bound. 3D: CL3D, CLVF



FSI embedded parameters

- **Structural influence domain**

- The contained fluid entities (centroids, interfaces) will be coupled
- Related to the size of the fluid mesh
- Too thin missing interfaces – leakages
- R : radius of influence sphere

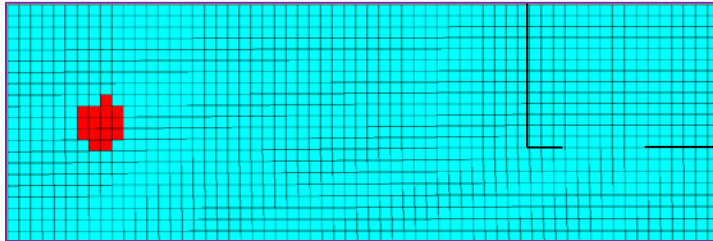
Prescribed (fixed) radius R of influence spheres at each coupled structural node. In the special, but frequent, case of a uniform structured fluid mesh (uniform square or cube elements) it is suggested to take R slightly larger than the semi-diagonal of a fluid element. This means that, for a 2D uniform square fluid mesh of side L_Φ one should take $R = 0.71L_\Phi$ while for a 3D uniform cube fluid mesh of side L_Φ one should take $R = 0.87L_\Phi$.

- **Fast search of coupled fluid entities**

- Speed up the calculation
- Essential for large models
- Minimum size + accuracy of results
- HGRI:
 - $h_g = 1.01 \max(h_F, h_S)$
 - h_F = size of the fluid mesh
 - h_S = size of the structural mesh

FSI Inputs

- Fluid only



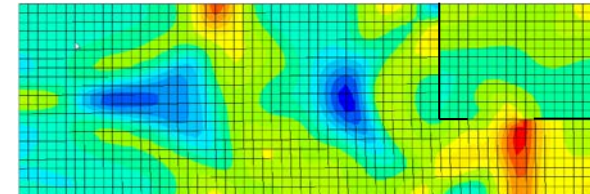
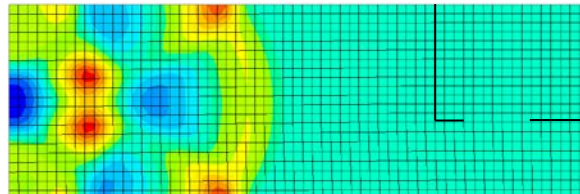
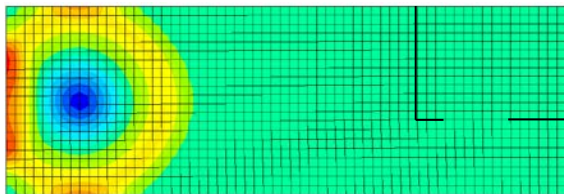
- High pressure zone
- Ambient pressure zone
- Duplicated nodes (reflecting boundaries)
- Free edges (reflecting boundaries)

```

2  ECHO
3  KFIL
4  DPLA EULE
5  GEOM Q4VF PART 1 PART 2 TERM

16 MATE GAZP RO 5.9485 GAMMA 1.4 CV 716.75
17      PINI 0.5E6 PREF 1.E5
18      LECT exp1 TERM
19      GAZP RO 1.1897 GAMMA 1.4 CV 716.75
20      PINI 1.E5 PREF 1.E5
21      LECT air TERM

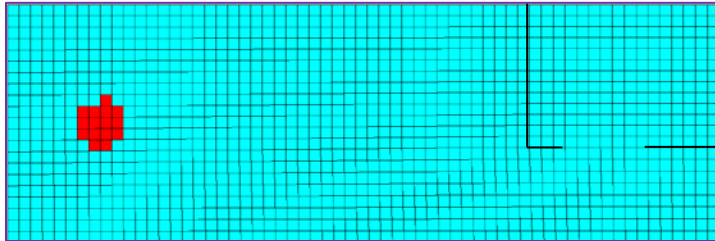
35 OPTI NOTE LOG 1
36 CSTA 0.5
37 VFCC FCON 6 ! Solveur HLLC
38 ORDR 2 ! Ordre 2 en espace
39 OTPS 2 ! Ordre 2 en temps
40 RECO 1 ! Reconstruction de type Green-Gauss
41 LMAS 3 ! k-limiteur de Dubois (eq. masse)
42 LQDM 3 ! k-limiteur de Dubois (eq. QDM)
43 LENE 3 ! k-limiteur de Dubois (eq. energie)
44 KMAS 0.75 ! Coefficient de limitation (eq. masse)
45 KQDM 0.75 ! Coefficient de limitation (eq. QDM)
46 KENE 0.75 ! Coefficient de limitation (eq. energie)
47 CENE ! Correction de l'energie interne
48 CALC TINI 0.0 TFIN 200.E-3
    
```



[Slides\videos\Ex2An1C.avi](#)

FSI Inputs

- Fluid + structure (conforming mesh)



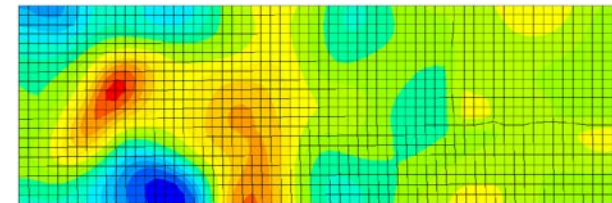
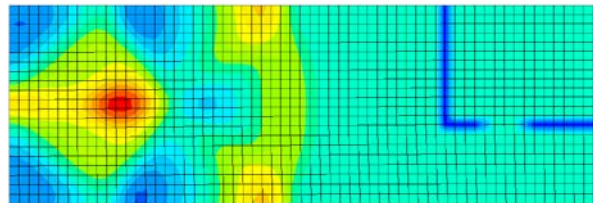
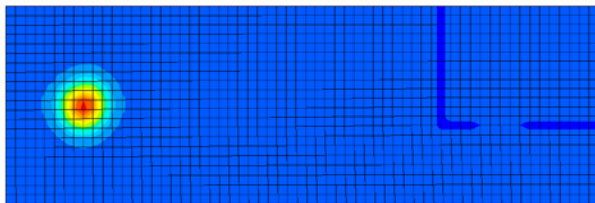
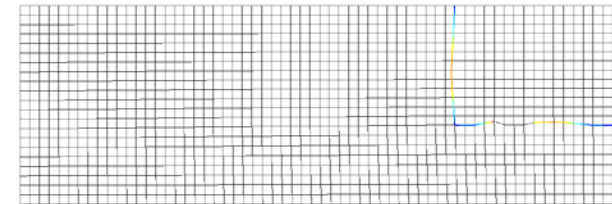
- High pressure zone
- Ambient pressure zone
- Structural elements (deformable)
- Free edges (reflecting boundaries)

```

2  ECHO
3  KFIL
4  DPLA ALE
5  DIME NALE 1 NBLE 1 TERM
6  GEOM Q4VF PART 1 PART 2 ED01 PART 3 PART 4 TERM

22 GRIL LAGR LECT NSET 5 TERM
23 EULE LECT NSET 6 TERM
24 AUTO AUTR

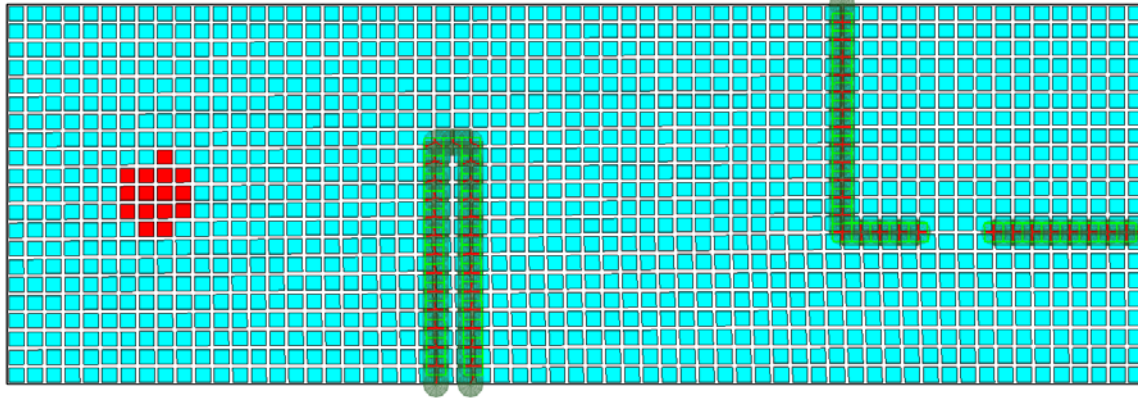
25 MATE GAZP RO 5.9485 GAMMA 1.4 CV 716.75
26     PINI 0.5E6 PREF 1.E5
27     LECT expl TERM
28     GAZP RO 1.1897 GAMMA 1.4 CV 716.75
29     PINI 1.E5 PREF 1.E5
30     LECT air TERM
31     VM23 RO 7800. YOUNG 1.6E11 NU 0.333 ELAS 1.05E8
32     TRAC 2 1.05E8 .656256E-3 1.6105E10 1.00066
33     LECT stem TERM
34 LINK COUP BLOQ 123 LECT NSET 7 TERM
    
```



[Slides\videos\Ex2An2.avi](#)

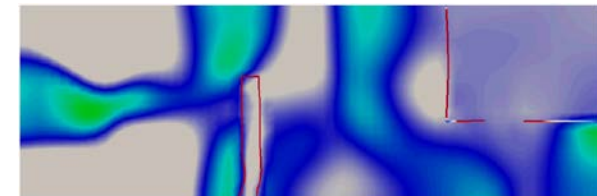
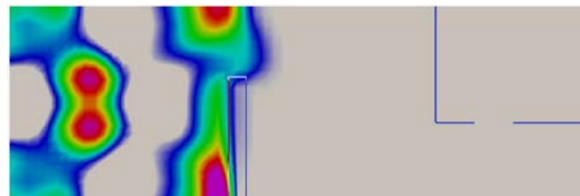
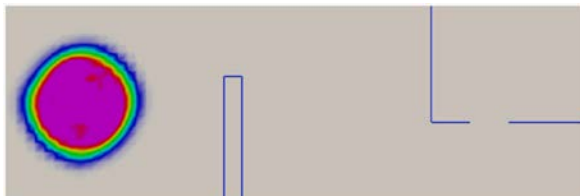
FSI Inputs

- Fluid + structure (embedded)



```
35 LINK COUP BLOQ 123 LECT NSET 9 TERM  
36 LINK DECO FLSW STRU LECT STRU TERM  
37 FLUI LECT flui TERM  
38 R 0.17  
39 HGRI 0.24  
40 DGRI  
41 FACE  
42 BFLU 2 FSCP 1
```

- FSI coupling
- Blocked fluxes



[Slides\videos\Ex2An3C.avi](#)

FSI conclusions

Model	Calculation time	Positive	Negative
EULER	8	Low calculation cost	No structure
FSA	51	Medium calculation cost Accurate	Rezoning No structural failure High preparation cost
FLSW	152	Medium preparation cost Accurate Significant deformations Structural failure	High calculation cost Sensitive to influence domain parameters

- **What approach should I select?**
 - The nature of the problem
 - Which phenomena should be considered

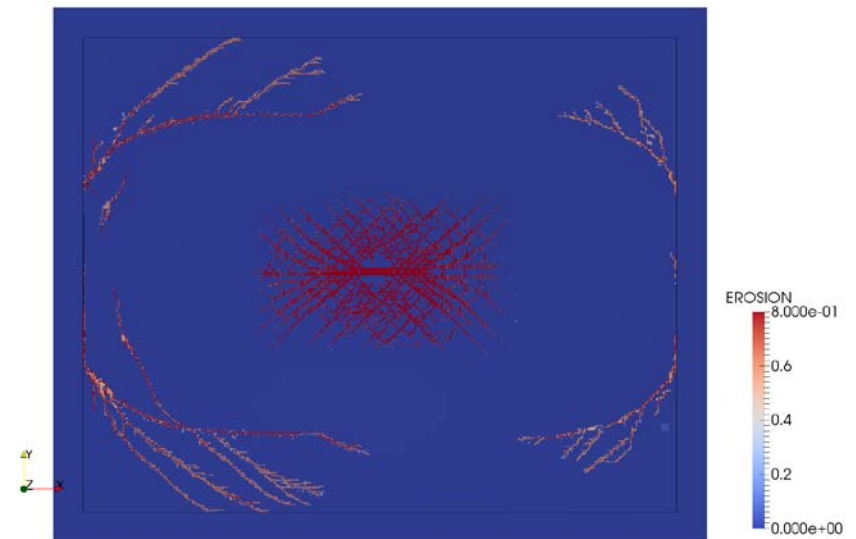
Failure / Erosion / Debris

Failure / erosion

- Failure: reaching a certain criteria in an IP
- Erosion: removing an element from the calculation

Failure: material behaviour:

- Damage
- Plastic strain
- Pressure
- ...



Failure (VM23)

```
"VM23"  "RO" rho  "YOUN" young  "NU" nu  "ELAS" sige ...  
        <"FAIL" $[ VMIS ; PEPS ; PRES ; PEPR ]$ "LIMI" limit>  
        "TRAC" npts*(sig eps)  
        /LECTURE/
```

VMIS Von Mises stress (isotropic criterion)

PEPS for a criterion based upon the principal strain

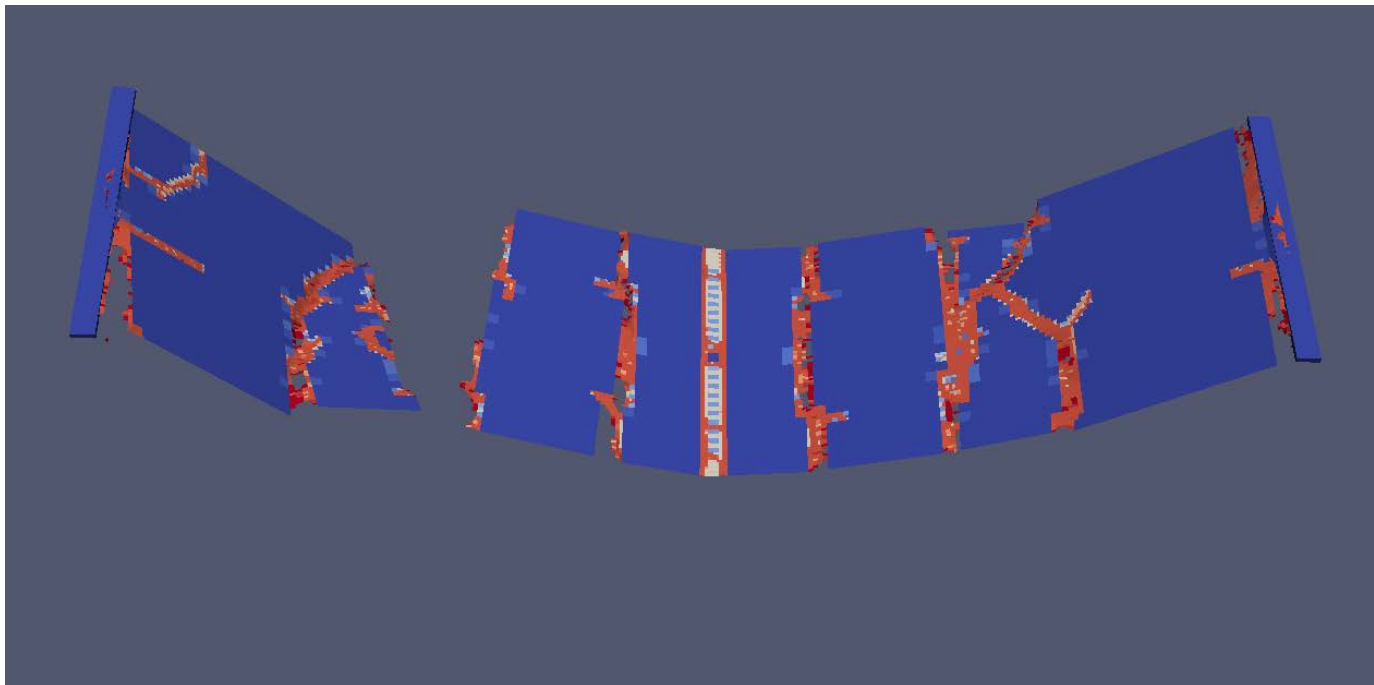
PRES for a criterion based upon the hydrostatic stress

PEPR for a criterion based upon the principal strain if the hydrostatic stress is positive (traction):

if the hydrostatic stress is negative (compression) there is no failure.

Erosion

- Material has reached a failure mode (damage or other criteria)
- element distorted (cannot be treated any more) (CROI)
- time step size is too small (CALC TFAI)
- Parts of the model should be removed at a certain time step or due to further criteria (displacement erosion, fantome elements)
- Erosion of attached element (CLxx)



Erosion

- EROS Idam CROI

Idam: ratio of failed IP to total IP from which on the element is eroded

- Options in the material law

- ParaView: VARI FAIL

- Threshold 0-Idam

```

1  EXO3
2  ECHO
3  CONV WIN
4  KFIL
5  EROS 0.25
6  TRID ALE
7  DIME
8  DEBR 5000
9  NALE 1 NBLE 1
10 TERM
11 GEOM Q4GS PART 1
12      T3GS PART 2
13 TERM
14 COMP
15      STFL VFCC X0 -5 Y0 -5 Z0 -5
16              LX 10 LY 35 LZ 10
17              NX 40 NY 140 NZ 40! 50*50=2500 elems and 51*51=2601
18              CLX1 CLX2 CLY1 CLY2 CLZ1 CLZ2

```

```

35 MATE
36 GAZP RO 1.3 GAMM 1.4 PINI 1.E5 PREF 1.E5
37 LECT flui DIFF hpres TERM
38 GAZP RO 65.0 GAMM 1.4 PINI 50.E5 PREF 1.E5
39 LECT hpres TERM
40 VPJC RO 7850.0 YOUN 2.1E11 NU 0.33 ELAS 3.7E8 mxit 500
41 QR1 2.364E8 CR1 39.3 QR2 4.081E8 CR2 4.5
42 PDOT 5.E-4 C 1.E-3 TQ 0.9 CP 452.0
43 TM 1800.0 M 0.0 DC 0.9 WC 473.0E6
44 LECT PART 1 PART 2 TERM
45 CLVF INFI RO 1.3 PRES 1.e5 GAMA 1.4 LECT abso TERM
46 LINK DECO
47 FLSW STRU LECT PART 1 PART 2 TERM
48 FLUI LECT flui TERM
49 R 0.2175 ! 0.87*H_FLUID = 0.87*0.5
50 HGRI 0.401 ! > THAN BIGGER STRUCTURAL ELEMENT !!!
51 DGRI
52 FACE
53 BFLU 2 ! BLOCK FLUXES
54 FSCP 1 ! COUPLE ALONG ALL DIRECTIONS
55 ECRI FICH PVTK TFREQ 0.2e-3
56 GROU AUTO
57 VARI ECRO DEPL VITE FAIL
58 OPTI NOTE
59 CSTA 0.5
60 LOG 1
61 FLS CUB8 2 ! to avoid problem with cub8 inverse mapping ...
62 VFCC FCONV 6 ORDR 2
63 CALC TINI 0 TEND 30.0E-3
64 FIN
65

```

Flying debris

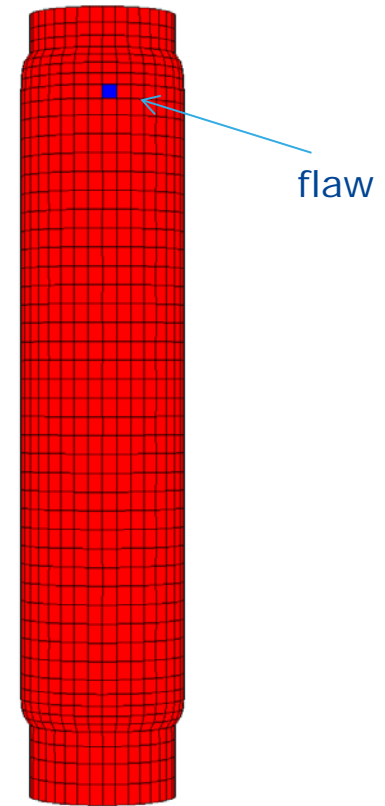
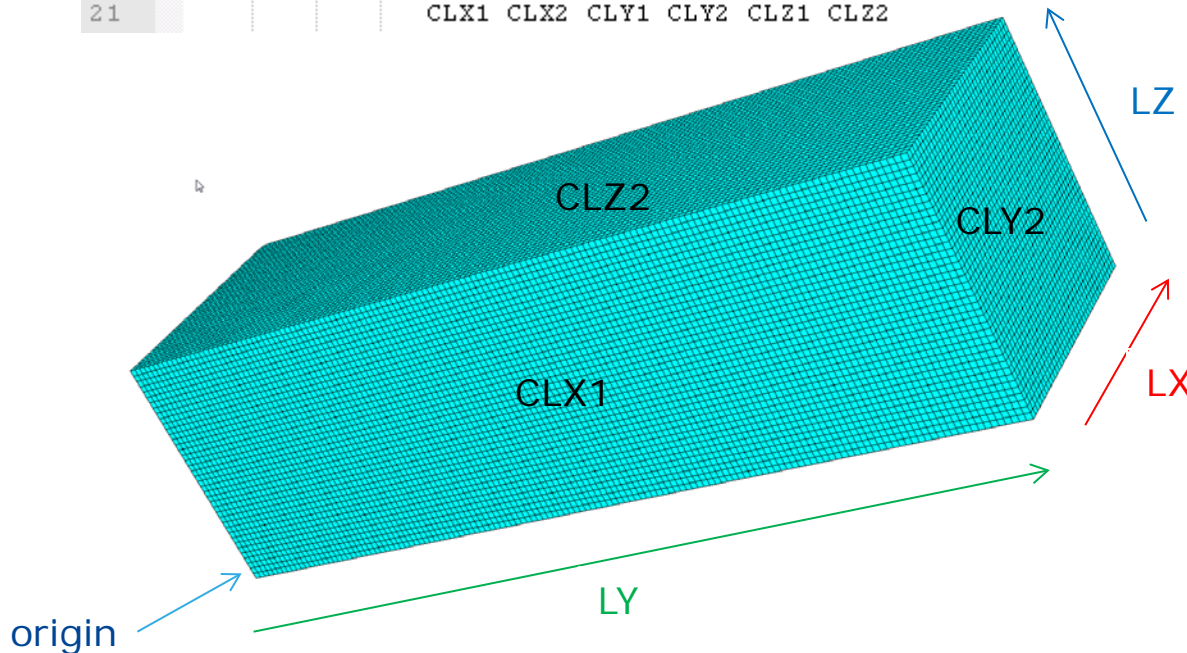
- Idea: material is in reality not eroded but mainly resulted in fragments
- Transferring the eroded elements in fragments
- Fragments were simple material points but could also be embedded in a fluid
- Drag forces and gravity could be added
- DEBR must be dimensioned at the beginning
- General debris parameters
- Creation of the debris

Example

FSI Inputs 3D explosion

- Explosion of tank in open field
 - Embedded FLSW
 - Flaw inserted via a thicker element
 - STFL fluid mesh construction

```
15  GEOM Q4GS PART 1
16  TERM
17  COMP
18      STFL VFCC XO -5 YO -5 ZO -5
19      LX 10 LY 35 LZ 10
20      NX 40 NY 140 NZ 40
21      CLX1 CLX2 CLY1 CLY2 CLZ1 CLZ2
```



FSI Inputs 3D explosion

- Explosion of tank in open field

Debris definition

```

9  DIME
10 DEBR 31176
    
```

Flaw definition

```

34 EPAI 0.01 LECT PART 1 DIFF flaw TERM
35 EPAI 0.0001 LECT flaw TERM
36 COUL TURQ LECT flui TERM
37 ROUG LECT PART 1 DIFF flaw TERM
38 VERT LECT abso TERM
39 BLEU LECT flaw TERM
40 JAUN LECT hpres TERM
    
```

```

71 DEBR
72 ROF 1.0 ! let particles move in vacuum
73 FLUI LECT flui TERM HGRI 0.251 ! grid size > fluid Smesh size
74 FILL PLEV 1 ! select the level
75 RO 2500 DRAG 1.0 COUP AFLY 1.0 OBJE LECT panels TERM
    
```

```

76 GRIL LAGR LECT buil _DEBR TERM
77 EULE LECT flui TERM
    
```

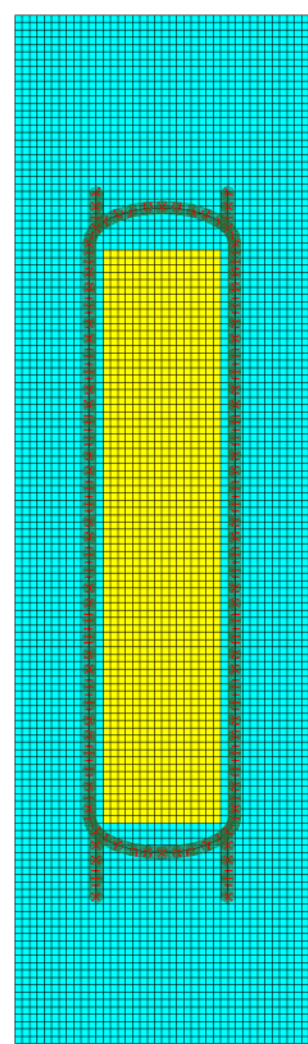
```

91 LINK DECO
92 FLSW STRU LECT PART 1 TERM
93 FLUI LECT flui TERM
94 R 0.2175 ! 0.87*H_FLUID = 0.87*0.25
95 HGRI 0.401 ! > THAN BIGGER STRUCTURAL ELEMENT !!!
96 DGRI
    
```

```

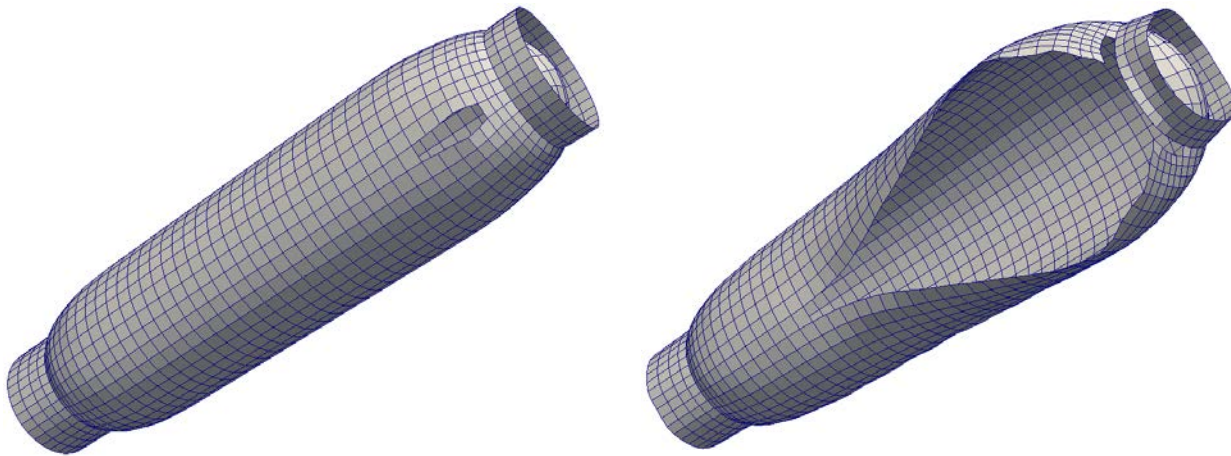
103 CHAR CONS GRAV 0.0 0.0 -9.81
104 LECT _DEBR TERM ! gravity acts only on debris particles
    
```

FSI parameters definition

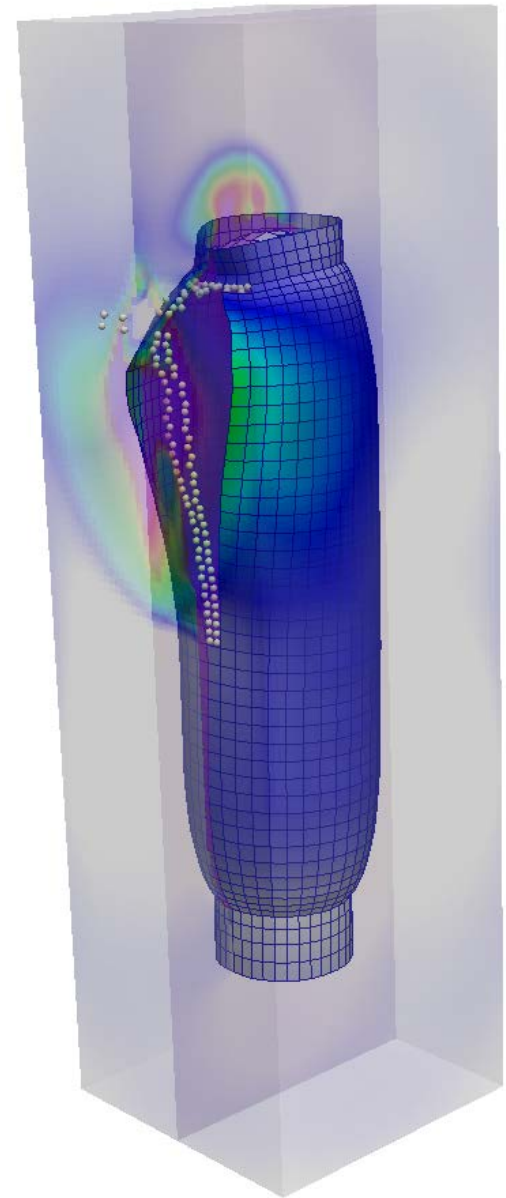


FSI Inputs 3D explosion

- Explosion of tank in open field
 - Results



- [Slides\videos\Ex3An1C.avi](#)



FSI Inputs 3D explosion

- Explosion of tank in open field + Adaptivity

```

9  DIME
10  DEBR 5000
11  ADAP NPOI 50000 CUVF 30000 NVFI 100000 CL3D 50
12  Q4GS 10000 ENDA
    
```

Adaptivity dimensioning

```

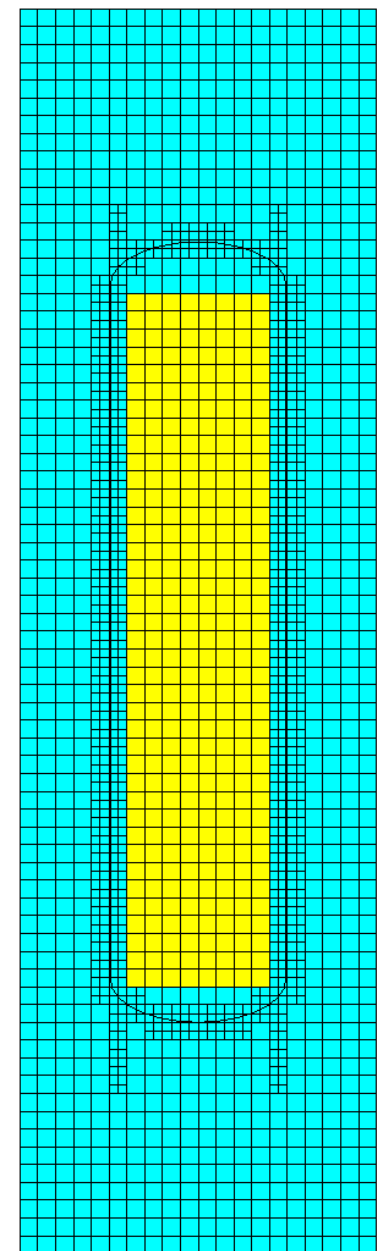
54  ADAP THRS ECRO 11 TMIN 0.45 TMAX 0.6 MAXL 3
55  LECT PART 1 TERM
    
```

Structural refinement parameters

```

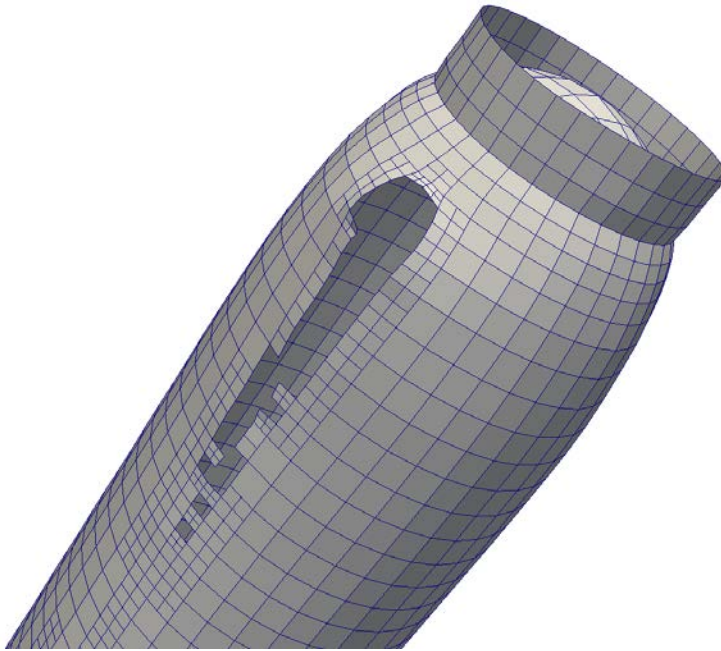
71  LINK DECO
72  FLSW STRU LECT PART 1 PART 2 TERM
73  FLUI LECT flui TERM
74  R 0.425 ! 0.87*H_FLUID = 0.87*0.5
75  HGRI 0.401 ! > THAN BIGGER STRUCTURAL ELEMENT !!!
76  DGRI
77  FACE
78  BFLU 2 ! BLOCK FLUXES
79  FSCP 1 ! COUPLE ALONG ALL DIRECTIONS
80  ADAP LMAX 2
    
```

FSI refinement parameters

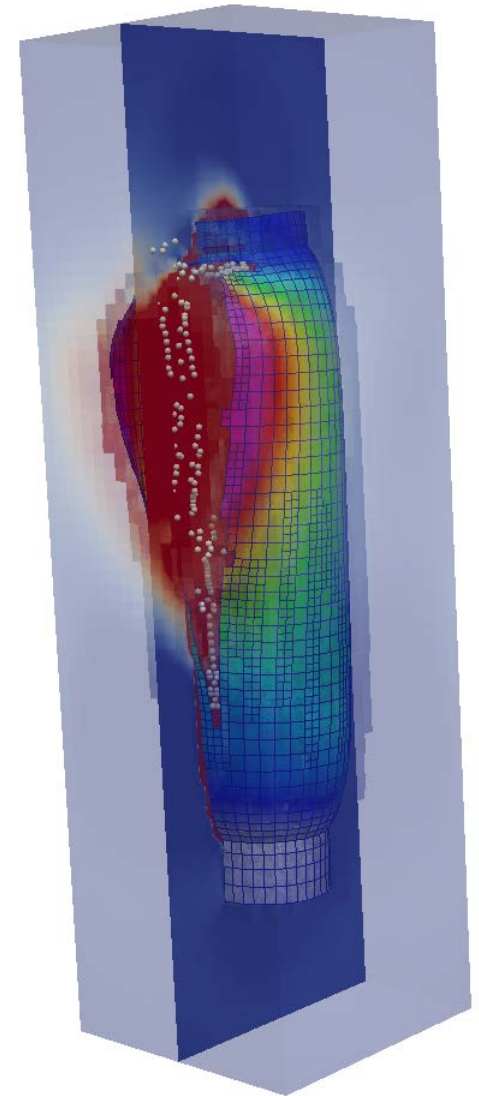


FSI Inputs 3D explosion

- Explosion of tank in open field + Adaptivity
 - Results



[Slides\videos\Ex3An2.avi](#)



Mesh Adaptivity

Mesh adaptivity

- Local refinement of the mesh
 - On some zones that considered as critical
 - Reduce the size of the model
 - High level of accuracy

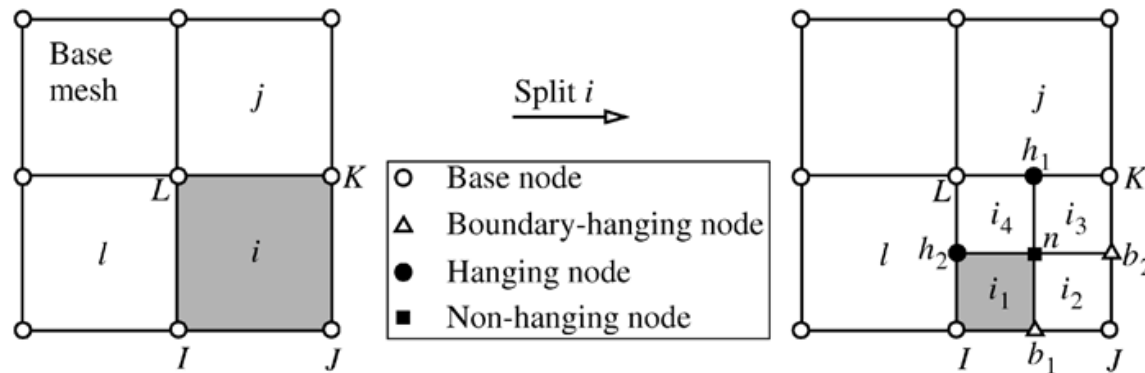


Fig. 1. Splitting a QUA4.

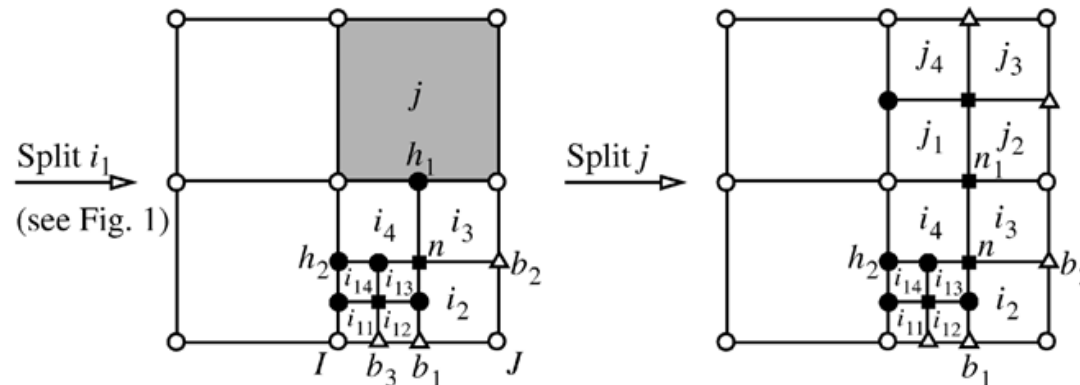
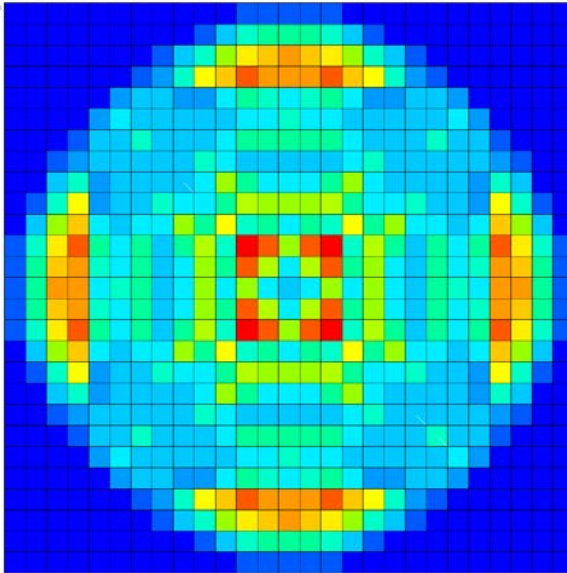


Fig. 2. Further splitting.

Structural mesh adaptivity

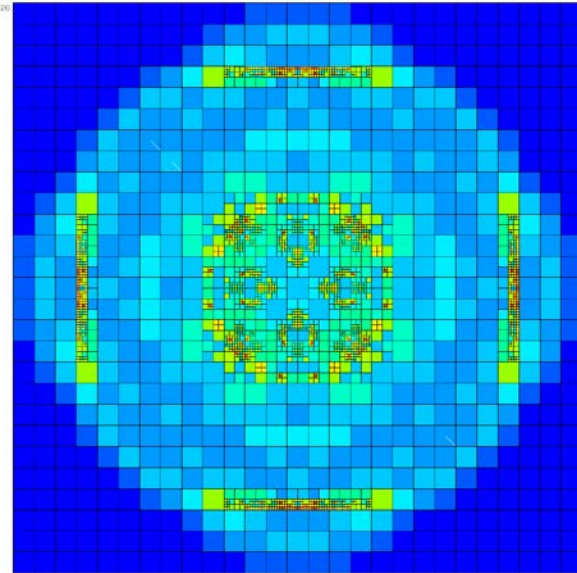
ADAP THRS ECRO 2 TMIN 50e6 TMAX 100e6 MAXL 4
LECT EROA TERM

PLATEHULL
TIME: 4.16330E-05 STEP: 20



MAX:19.00E+07
---149.49E+07
---148.75E+07
---148.75E+07
---147.50E+07
---147.50E+07
---146.25E+07
---146.25E+07
---145.00E+07
---144.75E+07
---144.75E+07
---143.50E+07
---143.50E+07
---142.25E+07
---142.25E+07
---141.00E+07
---141.00E+07
MIN:10.00E+00
EORG:2 [N+0]

PLATEHULL
TIME: 4.21560E-05 STEP: 20

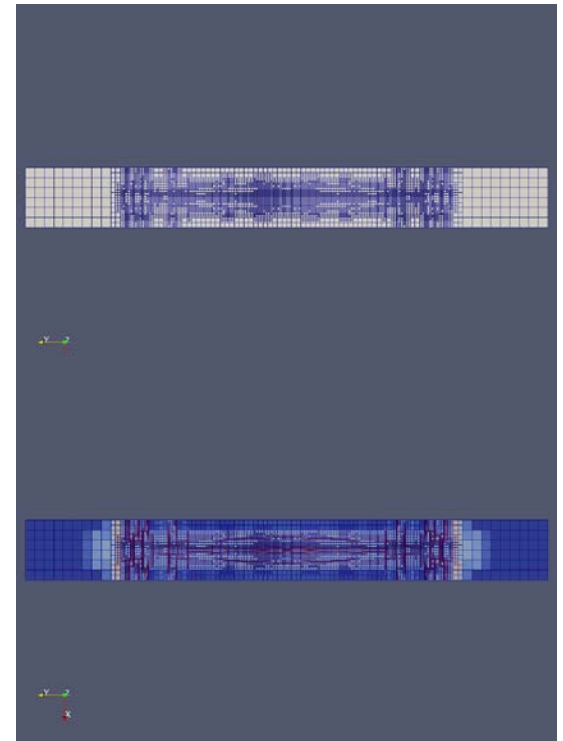
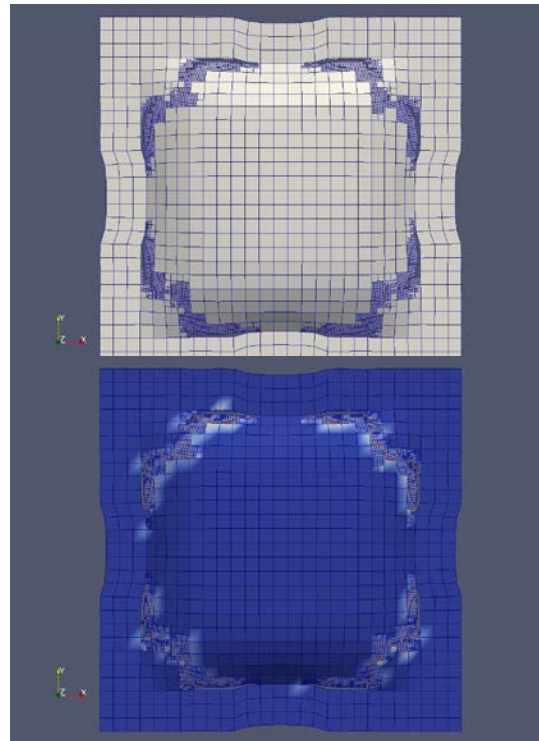
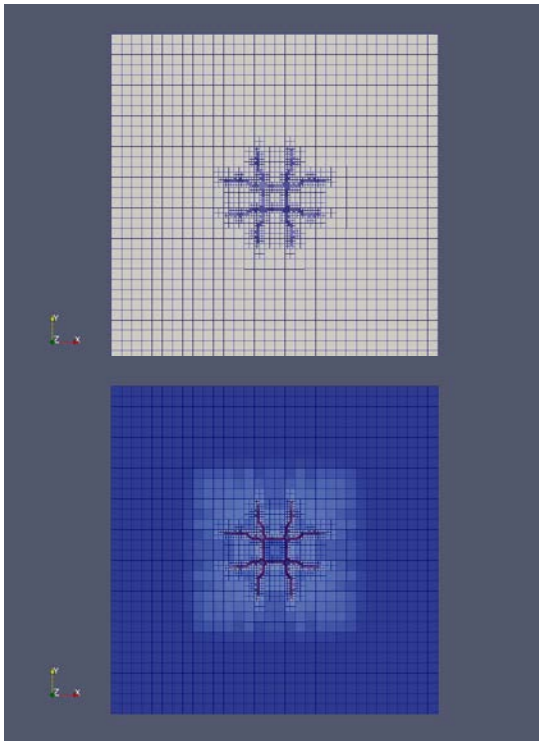


MAX:11.33E+08
---142.09E+08
---141.49E+08
---141.49E+08
---140.89E+08
---140.89E+08
---140.29E+08
---140.29E+08
---139.69E+08
---139.69E+08
---139.09E+08
---139.09E+08
---138.49E+08
---138.49E+08
---137.89E+08
---137.89E+08
MIN:11.70E+00
EORG:2 [N+0]

[Slides\videos\A0_Glis_AdapRev2.avi](#) [Slides\videos\BAF2.avi](#) [Slides\videos\mill04_join.avi](#)

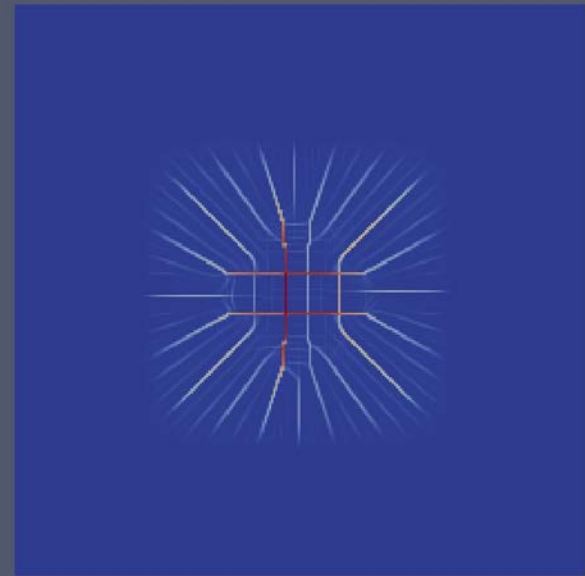
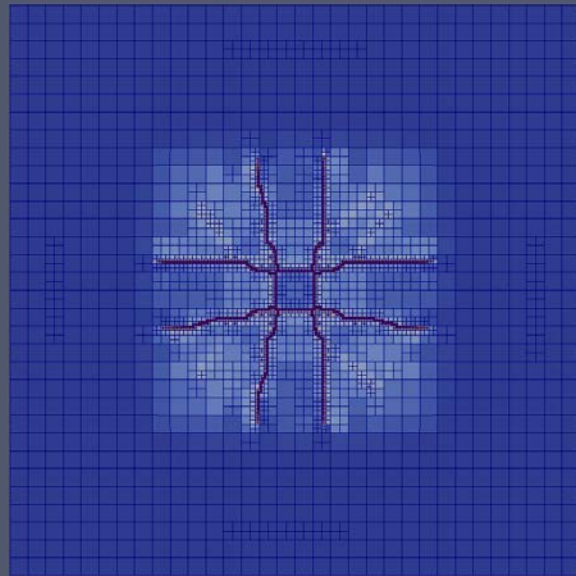
Structural mesh adaptivity

- Crack formation with damage driven adaptivity
 - Indicator should be a damage parameter
 - No unsplit since the damage parameter is a cumulative quantity

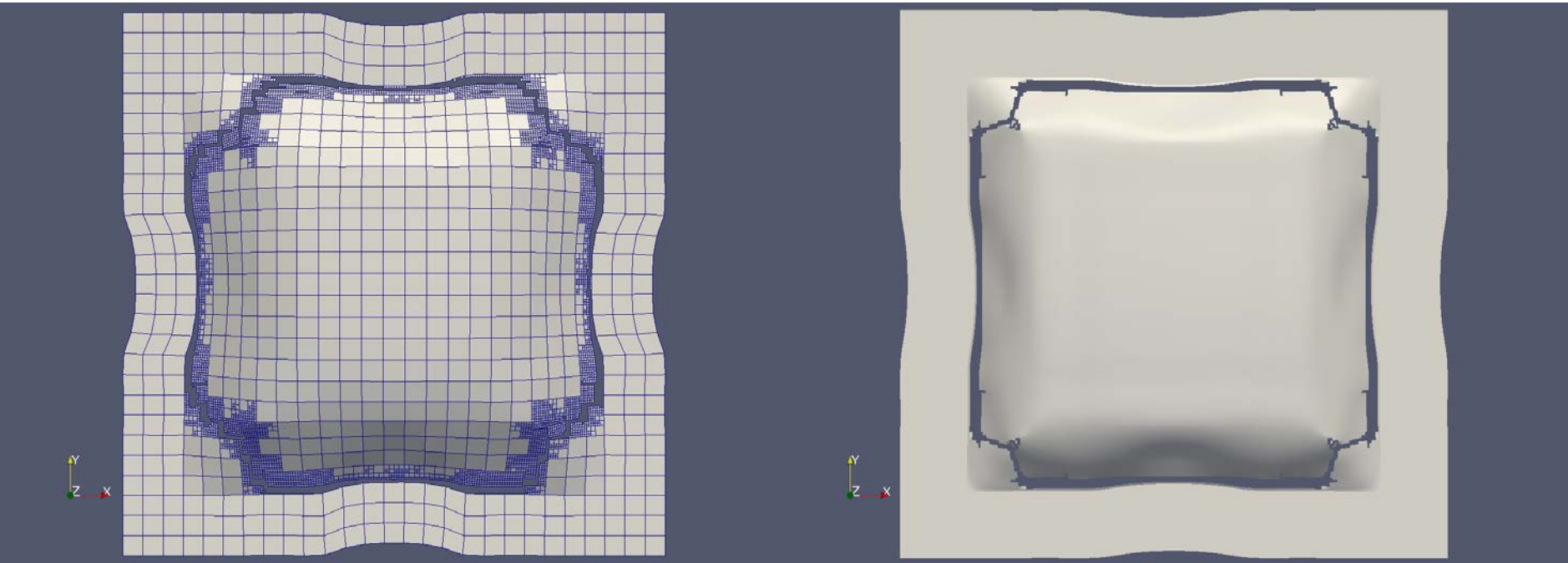


• [Slides\videos\A0_Glis_AdapRev2.avi](#) [Slides\videos\BAF2.avi](#)

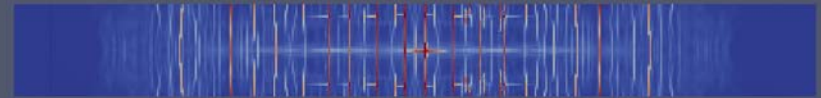
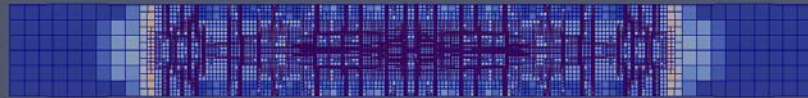
Structural mesh adaptivity



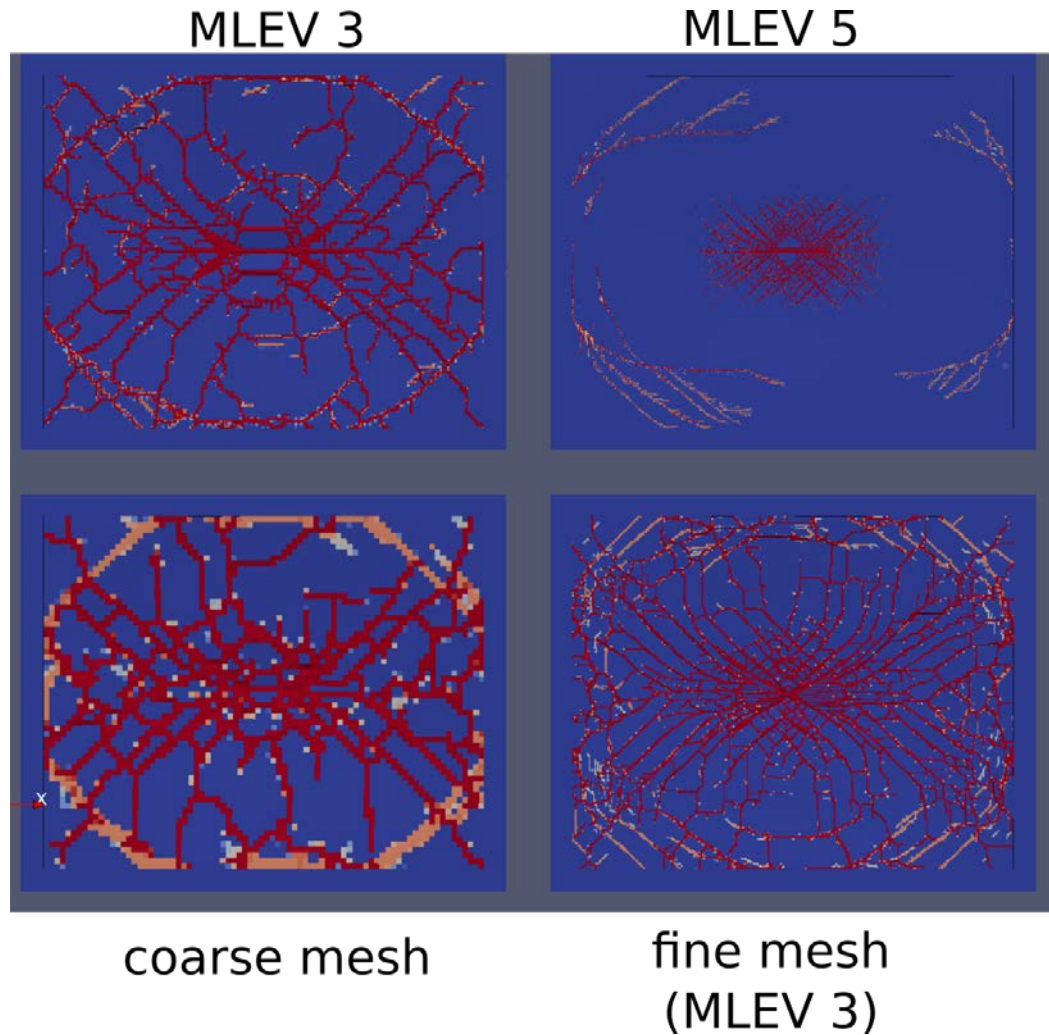
Structural mesh adaptivity



Structural mesh adaptivity

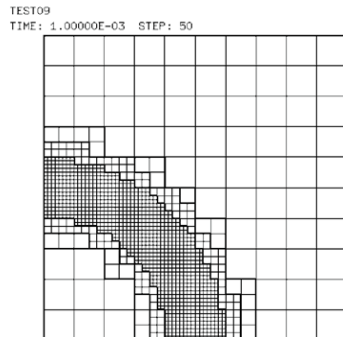


Structural mesh adaptivity

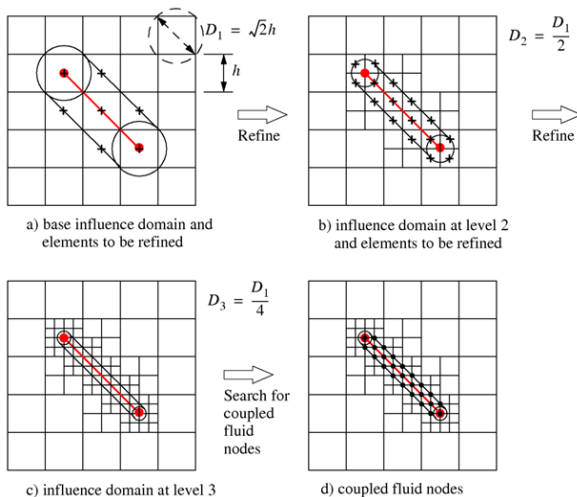


Fluid mesh adaptivity

- Indicator driven

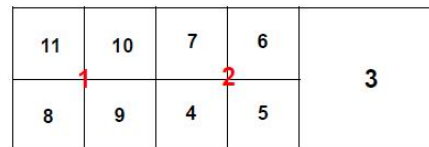
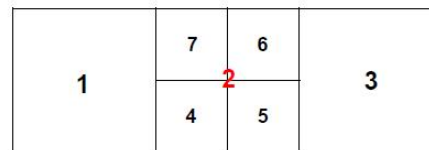
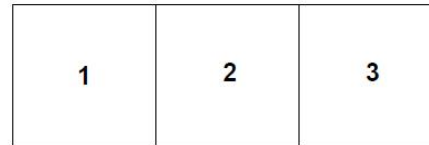


- FSI driven

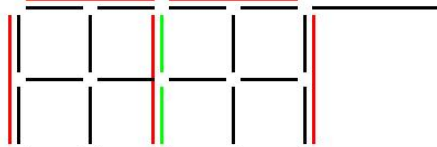
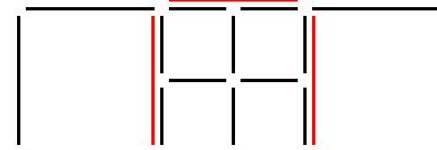
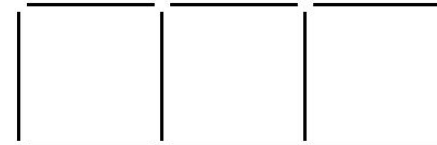


- Structure node
- + Element to be refined
- Coupled fluid node

Volumes



Interfaces



Base mesh
(refinement level 1)

Split central volume:
4 interfaces at level 1
become **inactive**
12 interfaces at level 2
are activated

Split left volume:
3 interfaces at level 1
become inactive
10 interfaces at level 2
are activated
2 interfaces at level 2
are **reused** (modified)

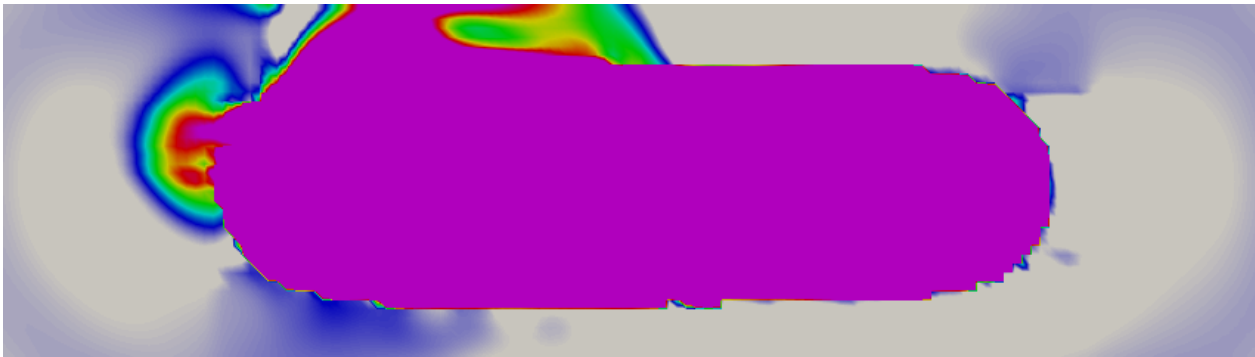
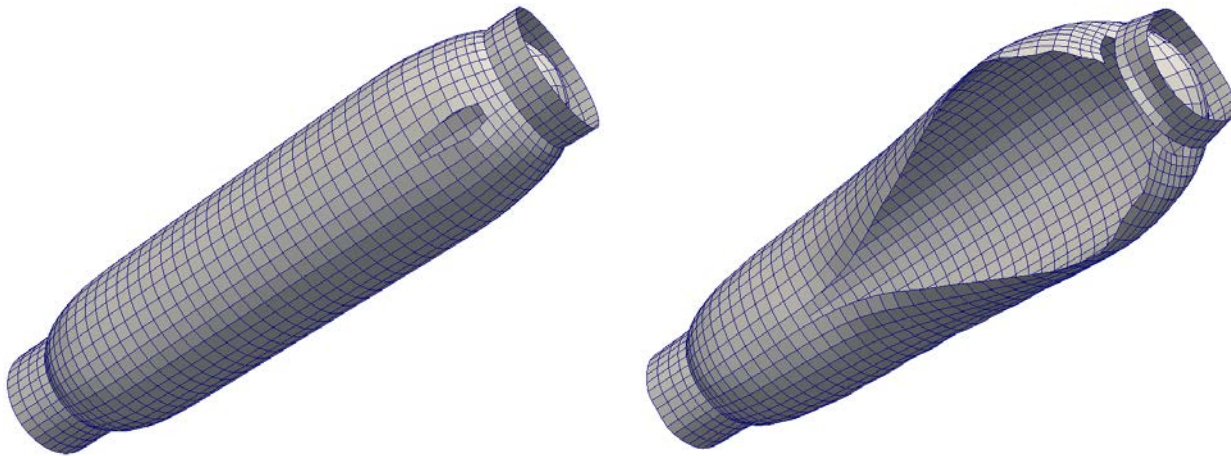
[Slides\videos\fullAdaPresavi.avi](#) [Slides\videos\mill04_join.avi](#) [Slides\videos\test_adap_3d.avi](#)

Mesh adaptivity

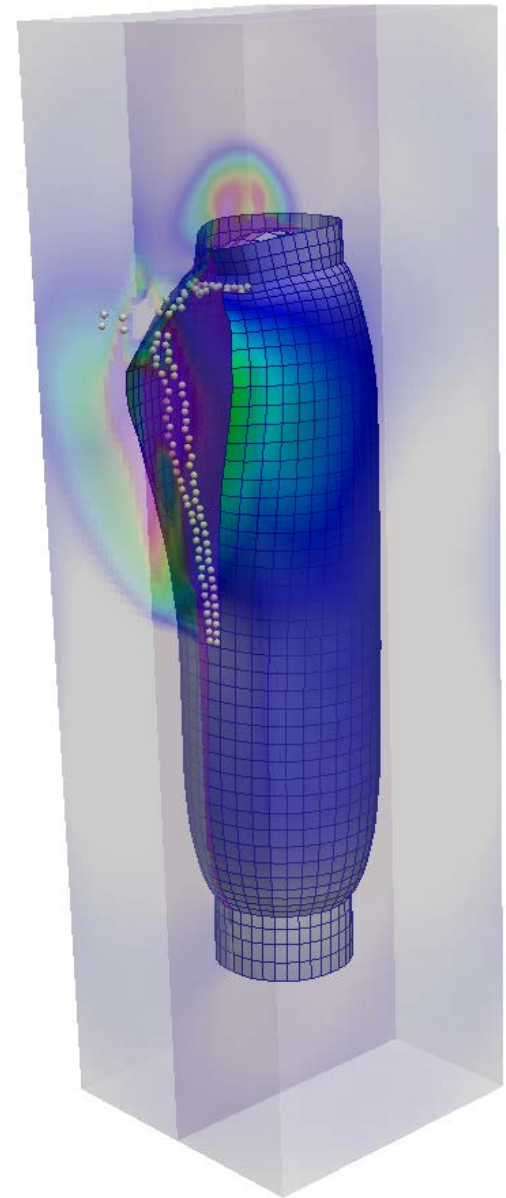
- Advantages
 - Less elements → less the calculation cost
 - Decrease the memory requirements for the calculation
 - The size of the output files can decrease significantly
 - We can have a period with higher time step

FSI Inputs 3D explosion

- Explosion of tank in open field
 - Results



- [Slides\videos\Ex3An1C.avi](#)



FSI Inputs 3D explosion

- Explosion of tank in open field + Adaptivity

```

9  DIME
10  DEBR 5000
11  ADAP NPOI 50000 CUVF 30000 NVFI 100000 CL3D 50
12  Q4GS 10000 ENDA
    
```

Adaptivity dimensioning

```

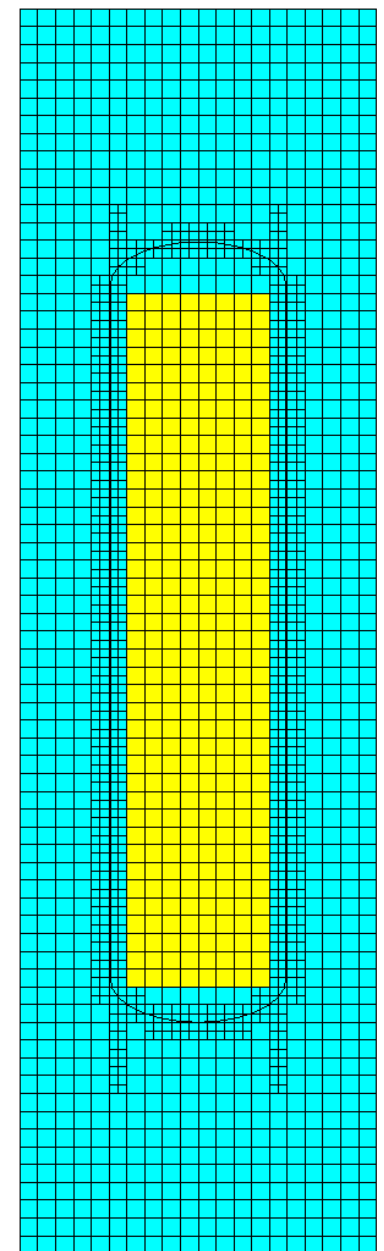
54  ADAP THRS ECRO 11 TMIN 0.45 TMAX 0.6 MAXL 3
55  LECT PART 1 TERM
    
```

Structural refinement parameters

```

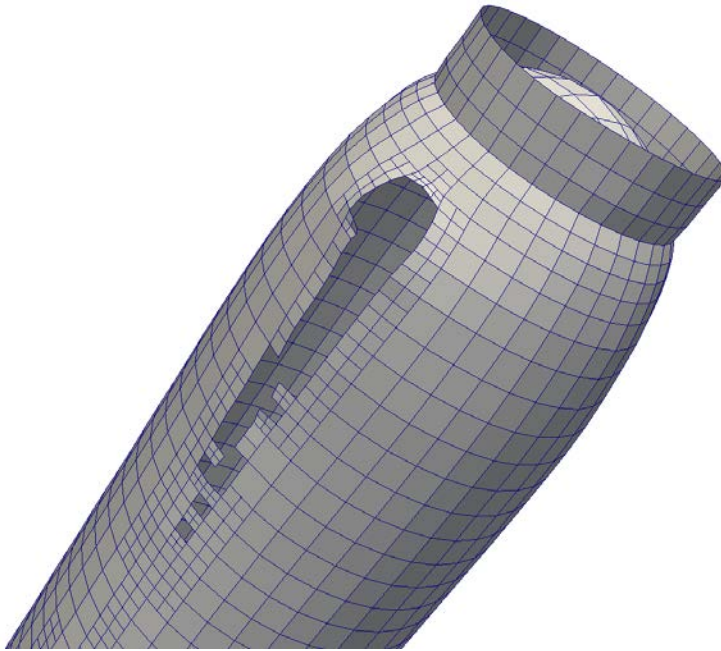
71  LINK DECO
72  FLSW STRU LECT PART 1 PART 2 TERM
73  FLUI LECT flui TERM
74  R 0.425 ! 0.87*H_FLUID = 0.87*0.5
75  HGRI 0.401 ! > THAN BIGGER STRUCTURAL ELEMENT !!!
76  DGRI
77  FACE
78  BFLU 2 ! BLOCK FLUXES
79  FSCP 1 ! COUPLE ALONG ALL DIRECTIONS
80  ADAP LMAX 2
    
```

FSI refinement parameters

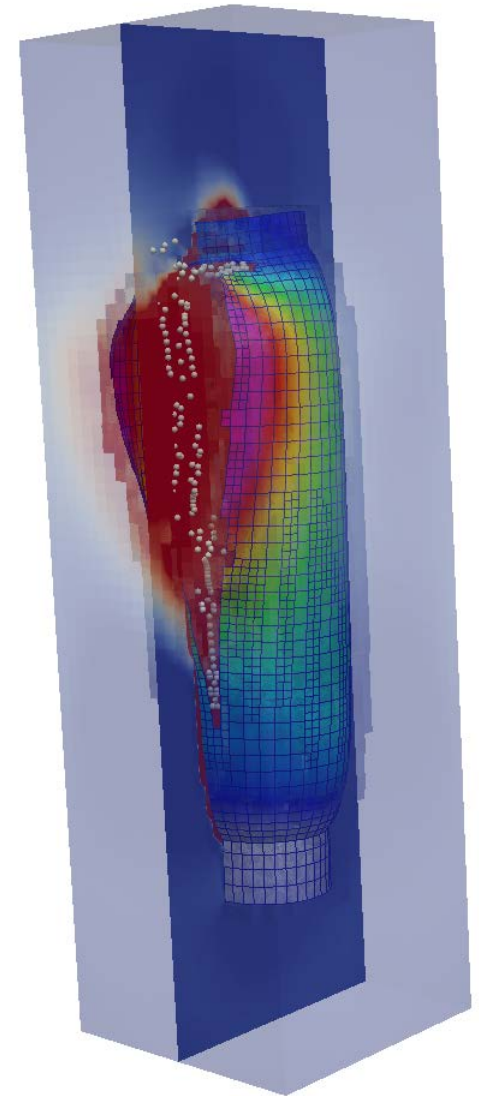


FSI Inputs 3D explosion

- Explosion of tank in open field + Adaptivity
 - Results



[Slides\videos\Ex3An2.avi](#)



Combustion

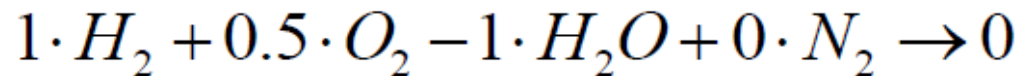
Combustion (burning)

- Chemical reaction between a fuel and oxidant
- Producing oxidized, often gaseous products

Explosions: Detonation – deflagration

- Reaction speed – speed of sound

Hydrogen explosions



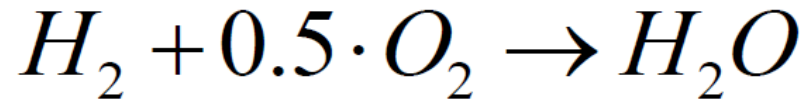
EUROPLEXUS: 2 material models GAZD and CDEM



Material models

- Euler equations (compressible, inviscid flow) for a mixture of perfect gases **in detonation regime**
- Chemical reaction represented by combustion of the hydrogen
- Associated kinetics taken into account
- Underlying equations: conservation of mass, momentum and energy, plus the equations of state of the materials and the relations describing the chemical reaction

Material models



After a certain delay interval dT (measured from the instant at which a certain critical temperature T is reached) combustion starts releasing a certain amount of energy q into the system

Material models – GAZD \leftrightarrow CDEM

- GAZD specific for detonation
- CDEM can be used for a wider range (strong detonation to weak deflagration)
- CDEM More general
- CDEM More expensive
- Course concentrates on CDEM

Material models CDEM

- At least two zones are needed: burnt and unburnt
- Each definition has two parts: general and components

```
MATE CDEM PINI 32.E5 ! Zone 1 (burnt)
      PREF 1.E5
      TINI 3000.
      KSI0 0.999
      K0 20000. ! Unphysically high value: the code will evaluate
                ! and use the theoretical detonation speed instead
      TMAX 6000.
      R 8.314
      NESP 4
      ORDP 4
      NLHS 2
```

General

```
COMP1 ! H2
      MMOL 2.E-3 H0 -4.195E6 CREA 1.0
      CV0 9.834E3 CV1 5.427E-1 CV2 8.622E-4
      CV3 -2.373E-07 CV4 1.847E-11
      YMAS 2.850E-02

COMP2 ! O2
      MMOL 32.E-3 H0 -2.634E5 CREA 0.5
      CV0 5.750E2 CV1 3.505E-1 CV2 -1.283E-4
      CV3 2.336E-8 CV4 -1.533E-12
      YMAS 2.264E-1

COMP3 ! H2O
      MMOL 18.E-3 H0 -1.395D7 CREA -1.0
      CV0 1.156E3 CV1 7.683E-1 CV2 -5.731E-5
      F4E_Examples.doc 01/03/15 @ 11:23
      Page 43 of 95
      CV3 -1.827E-8 CV4 2.445E-12
      YMAS 1.274E-11

COMP4 ! N2
      MMOL 28.E-3 H0 -2.953D5 CREA 0.0
      CV0 6.529E2 CV1 2.882E-1 CV2 -7.804E-5
      CV3 8.782E-9 CV4 -3.055E-13
      YMAS 7.451E-1 ! Sum of YMAS must be 1.0
      UCDS 1 ! Limited reconstruction + UCDS

LECT expl TERM
```

Components

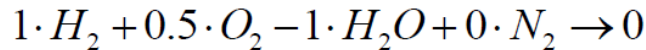
Material models CDEM

- PINI: pressure
- PREF: reference pressure (1 bar)
- TINI: initial temperature
- KSIO: Initial volume fraction, number close to 1 but not 1.0
- KO: flame speed, chosen high, theoretical value is used
- TMAX: max temperature at which the polynomial expression giving the heat capacity at constant volume
- R: gas constant (8.314 J/(mol K))
- NESP: number of components
- ORDP: order of the polynomial equation
- NLHS: number of reactants (H2 and O2)

```
MATE CDEM PINI 32.E5 ! Zone 1 (burnt)
      PREF 1.E5
      TINI 3000.
      KSIO 0.999
      KO 20000. ! Unphysically high value: the code will evaluate
                ! and use the theoretical detonation speed instead
      TMAX 6000.
      R 8.314
      NESP 4
      ORDP 4
      NLHS 2
```

Material models CDEM

- MMOL: molar weight in kg/mol
- H0: enthalpy of formation in J/kg at T=0K
- CREA: coefficient in the reaction equation: positive for reactants, negative for products and 0 if inert



- CV0-CV4: polynomial specific heat

$$C_V(T) = C_{V0} + C_{V1}T + C_{V2}T^2 + \dots + C_{V\text{ordp}}T^{\text{ordp}}$$

- YMAS: initial mass fraction

```

COMP1 ! H2
MMOL 2.E-3 H0 -4.195E6 CREA 1.0
CV0 9.834E3 CV1 5.427E-1 CV2 8.622E-4
CV3 -2.373E-07 CV4 1.847E-11
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MMOL 32.E-3 H0 -2.634E5 CREA 0.5
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CV3 2.336E-8 CV4 -1.533E-12
YMAS 2.264E-1

COMP3 ! H2O
MMOL 18.E-3 H0 -1.395D7 CREA -1.0
CV0 1.156E3 CV1 7.683E-1 CV2 -5.731E-5
F4E_Examples.doc 01/03/15 @ 11:23
Page 43 of 95
CV3 -1.827E-8 CV4 2.445E-12
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COMP4 ! N2
MMOL 28.E-3 H0 -2.953D5 CREA 0.0
CV0 6.529E2 CV1 2.882E-1 CV2 -7.804E-5
CV3 8.782E-9 CV4 -3.055E-13
YMAS 7.451E-1 ! Sum of YMAS must be 1.0
UCDS 1 ! Limited reconstruction + UCDS
    
```

Material models CDEM: output (ECRO)

Numbering very difficult – numbering written in the listing, for 3D:

- **1: pressure of the mixture**
- **2: density of the mixture**
- **3: maximum sound speed**
- **4: Volume fraction of unburnt**
- 5: density of unburnt
- 6: x-velocity of unburnt
- 7: y-velocity of unburnt
- 8: z-velocity of unburnt
- 9: pressure of unburnt
- **10: Volume fraction of burnt**
- 11: density of burnt
- 12: x-velocity of burnt
- 13: y-velocity of burnt
- 14: z-velocity of burnt
- 15: pressure of burnt
- 22: FUNDAMENTAL FLAME SPEED
- 23: ABSOLUTE TEMPERATURE OF THE MIXTURE

...

Options

Several options

- In particular for finite volume solution

```
OPTI NOTE LOG 1
CSTA 0.1
VFCC RECO 3 ! Green-Gauss for CDEM/DEMS
      ERK2  ! Explicit Runge-Kutta second order time integration
      LMAS 2 ! Barth-Jespersen limiter for mass conservation
      KEAR 1 ! K is defined by barycentric coordinates
      PAS0 1.0D-12 ! Deltat at zero step
```

Easy input (!)

Definition of CDEM material only once for all parts
Some values were overwritten later on
The different parts are defined with the INIT

```
INIT ! Now we fine-tune the initial conditions
! for the CDEM material
VFCC ! Initial conditions for the burnt phase
  VITX 0.0
  VITY 0.0
  VITZ 0.0
  PINI 32.E5
  TINI 3000.0
  KSI0 0.999
  K0 20000. ! Unphysically high value: the code will evaluate
            ! and use the theoretical detonation speed instead.
  Y1 2.850E-02
  Y2 2.264E-1
  Y3 1.274E-11
  Y4 7.451E-1
  LECT H2_burnt TERM ! "burnt" is the name of the mesh zone
                    ! initially containing (mostly) the burnt phase
```


Examples

